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> MEASUREMENT OF SCAVENGING EFFICI-ENCY OF THE TWO STROKE ENGINE:A COMPARISON AND ANALYSIS OF METHODS

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Thesis I6



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Dear Sir:

In partial fulfilment of the requirement for the degree of Navel enrineer, from the Massachusetts Institute of Technology, we hereby submit our thesis entitled: Measurement of Scavencing Officiency of the 2-Stroke Engine: Analysis and Comparison of Nethods.

Respectfully,



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The formula i.h.p. = ρ_8 N V_D $\frac{r}{r-1}$ e_s F/A E_c η_1 (symbols defined below) denonctrates the relation between the indicated horsenower of the two stroke engine and the engine operating variables.

Os = scavenging density: air density at inlet temperature and exhaust pressure - lbs/ft3

N = Power strokes per minute

 V_D = displacement volume - piston area x stroke in ft³

r - compression ratio - non dimensional

F/A = lbs fuel/lbs air - based on air retained in the cylinder - fuel-air ratio

E_c = Heating value of fuel - btu/15 fuel - 18,900 for 100 octane

1 = Overall indicated thermal efficiency - non dimensional
i.h.p. = indicated horsepower of two stroke engine

es is defined as the ratio

air retained in cyl. in lbs/min

or N V r (Air, of density ps, required to rel fill cylinder in lbs/min.)

and is known as scavenging efficiency.

R_s is defined as the ratio $\begin{bmatrix} air supplied to engine, in 1bs/min \\ \rho s & V_D \frac{r}{r-1} \end{bmatrix}$

and is known as scavenging ratio.

It can be seen from the above information that, all other veriables being equal, the ihp of the two stroke is proportional to es. It follows that an accurate means of determining es is



a necessity.

The object of this thesis has been to establish comparative information on the accuracy of four accredited methods for measuring this quantity.

A corollary to the main work was the design and tost of a sampling check valve. The general characteristics of the valve were suggested by Profs. A. R. Rogowski and C. F. Taylor. Fig. XIX shows photographs of completed design. The methods were run simultaneously on an engine in steady state conditions, defining four curves (one per method) for each speed run. Runs were made at 1000 and 1400 RPM, and at each speed, R_S was varied from 1.0 to 1.8, in six steps. The engine used in the evaluation was a single cylinder two-stroke, loop-scavenged spark ignition, Waukesha CFR type.

The cylinder is an M.I.T. design, for two stroke operation mounted on the Waukesha CFR crankcase. Briefly, the methods employed were as follows:

Tracer Gas Method - Hereafter known as Method I.

Monomethylamine gas was injected into the inlet air stream, giving concentrations of from 1 to 2% by volume, of monomethylamine. The monomethylamine gas in the air retained in the cylinder dissociated in the combustion chamber, under the heat of burning while gas in bypassed air remains unaffected. By measuring the concentrations of monomethylamine in the inlet and exhaust streams, direct measures of the ratio of the amount of air entering engine, to the air not retained in cylinder were obtained.



Application of corrections for burning efficiency and shrinkage, and use of calculations as outlined by Schweitzer and DeLuca in NACA Technical Notes #838 led to determination of eg. Details of procedure and calculation in Appendix A.

Gas Analysis of Compression and Expansion Samples - Method II.

A sampling valve capable of drawing samples from any point in the cycle was used in conjunction with an Orsat analysis, to analyze a compression, and an expansion sample. The Cox valve used for sampling occupied 8 and 11 degrees crank angle at speeds of 1000 and 1400 RPM, respectively, from "start of open," to "fully closed." An oxygen content balance was used, with straightforward weight calculations, to evaluate es. The Gerrish and Neems chart and the D'Alleva and Lovell chart of "Analysis of Exhaust Gases of 4 stroke Engine" were used to evaluate molecular weight of residuals. Isentropic expansion and compression were assumed in the region of port openings on expansion through the early part of compression, for the evaluation of temperatures accompanying samples. An arbitrary 300°F temperature drop due to "blowdown" heat exchange was assumed. Samples for all runs were drawn at crank angles of 1110 (just prior to opening of exhaust ports) and 3090 (about 600 after exhaust ports close, on compression stroke).

Details of procedure and calculations in Appendix A.

Gas Analysis of Expansion Sample - Method III.

The results of analysis of the residual gas sample described in Method II were used to establish a vertical line on each of



the two Exhaust Gas Analysis Charts. The ordinates of the charts were "% by volume" of the elements and compounds in in residual gases plotted arainst fuel-air ratio as abscissae. The fuel-air ratios defined by the vertical lines established as above, were used as the fuel-air ratio, under the particular conditions, and a single calculation produced es. Details in Appendix A.

Isac Method (Indicated specific air consumption) - Method IV.

Sloan Laboratory made available a 4-stroke CFR type singlecylinder spark ignition engine of the same bore and stroke as
the two-stroke used in the thesis. This 4-stroke engine was
run at the same piston speed, jacket water outlet temperature,
inlet air temperature, and compression ratio as the two stroke,
for speeds of 1400 and 1000 RFM. Best power fuel-air ratio was
used (.078). Indicator cards, and air and fuel data were taken.
Comparison of the 2 stroke and 4 stroke data in conjunction with
corrections applied on a calculated basis gave an approximation of the indicated specific air consumption of the two stroke
engine. This figure combined with the i.h.p. as taken from the
two-stroke card yielded "air retained," thence eg. Details in
Appendix A.

The sampling check valve was tested to determine its suitability as an alternative for the electrically operated timed sampling valve, as used in Methods II and III. It was mounted in the exhaust ports as shown in Fig. XIV, and could be used to take exhaust samples only.



General Information

The Waukesha engine used in this work was run at best power settings for the particular operating conditions. This was done for two reasons: First, it permitted the use of the Isac Method which requires comparison of 2-stroke and 4-stroke engines at, or near, best power donditions, and second, because the operation of the Waukesha was steady at best power conditions.

Results

Smooth curves of $e_s vs R_s$ were obtained from Methods II, III, and IV over the range $R_s = 1.0$ to $R_s = 1.8$ at spieds of 1000 and 1400 RPM. The curves of $e_s vs R_s$ for Method I were the least predictable from results obtained. Method I was the most troublesome of all used, and any estimate of its value must be qualified. These qualifications are enlarged on, in appropriate sections.

The early tracer gas method results were extremely bad, due to burning and dissociation of monomethylamine outside cylinder, and loss of monomethylamine to condensation liquids in exhaust tank. It was not until a method was devised of picking up the exhaust stream samples just outside the ports, that tracer gas method results improved. The first gas samples taken with the Cox sampler gave erratic results. After a system was set up for constantly checking the valve for the smallest leak, results steaded down. Technique improvement on the Crsat



gas analysis equipment led to reliable results soon after first runs were made. Methods II and III were both extremely sensitive to the Orsat analysis results.

The final curves of es vs Rs showed surprisingly small discrepancies among the methods. The differences between the results of any two methods in es at the same Rs on the curves were never more than .061 and averaged close to .04. The curves exhibited similar tendencies in slope, curvature, and intersection. At Rs = 1.0, curves of 1400 RPM intersected curves of 1000 RFM, for Methods I, II, and III. Above Rs = 1.0, 1400 RPM curves showed higher absolute values of es than 1000 RFM curves. Taking any single curve (1000 or 1400 RPM) the "Spreads" in es found by making successive runs at a single operating condition (vertical "spread") was (a) for Method I, about .10 (b) for Method II, about .05, (c) for Method III about .2%, (d) no "apread" was measured in Method IV since only one run was made or calculated at each condition.

Note that the results obtained in three of the four methods depend directly or indirectly on the performance and location of the timed sampling valve. This effect can be evaluated in sample calculation sections.

The results of the various methods compared more favorably with each other, than expected by the authors. Data figures produced the results toward which all estimates, and previous results pointed. In order to get what were considered good



results, numerous check runs were made. The results of the check runs were always an improvement, indicating that lack of precision in earlier runs was almost entirely due to either faulty technique, or, in the case of Methods II and III, small leakage in the sampling valve. The assumptions made at various steps in the methods were arbitrary, and in some cases were almost matters of individual opinion, but the end results were satisfactory nonetheless, for the effects of the assumptions were small, and they served the purpose of permitting the methods to be evaluated, using simple, rapid calculations.

The sampling check valve tested successfully for five hours running time. It gave the most satisfactory gas analysis results, using the Orsat equipment, and on the basis of those results, surpassed the timed sampling valve. At the end of five hours time, it stopped operating, and examination revealed that it had clogged with residue from exhaust gases. Enough values for one es vs Rs curve, at 1000 RPM were obtained. Time precluded further investigation.

Conclusions

The one solid conclusion that can be drawn from this work is that all the methods will give the same quantitative answer within 10 percent "precision," where "precision" is used to describe a variation, and not absolute values.

Having drawn this conclusion, facts pertinent to the advantages or disadvantages of each method may be added to permit



evaluation by interested parties.

For <u>overall</u> use, precision, speed, facility, simplicity, etc., Method III is chosen as most satisfactory.

The necessity for an expensive, bulky, complicated sampling valve, and the slow Orsat analysis equipment is the only detraction from this choice.

Method II gave good results but demands time, several cumbersome calculations and the use of an Orsat and sampling valve. The valve used in these runs and in Method III is expensive, and extremely sensitive to small leakage.

Method I gave acceptable results over a range from $R_s = 1.2$ to $R_s = 1.65$. At $R_s = 1.0$ and $R_s = 1.80$, unexplained poor results were obtained. This method gave the most erratic values of e_s . The adaptation of this method to spark ignition engines requires a more complete trial than this program devoted to the subject. It is felt that the method has merit, and can be made reliable.

Method IV gave the most satisfactory results on a basis of precision, and a smooth eg vs Rg curve. The necessity for a comparable 4 stroke engine in addition to the 2 stroke being examined, and for indicating equipment limits its use to properly equipped laboratories. It is time-consuming and cumbersome because of indicator card requirements.

The sampling check valve, of the type tested, will make an acceptable substitute for the expensive, complicated, electrically operated timed sampling valve, for use with Method III. Design



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changes are required to offset clogging of valve, and further testing seems well worthwhile.

Recommendations

- A. For future work in this direction, it is recommended that an investigation be made of the practicability of injecting into the cylinder, after the ports have closed on the compression stroke, an inert substance which is (1) not contained in residuals or fresh charge in any quantity, (2) not decomposed by combustion temperatures, (3) may be accurately measured for its concentration in the exhaust stream by some method perhaps similar to the tracer gas concentration measurement, and (4) lends itself to injection against about 600 psia.
- B. It is also recommended that technique in handling and familiarity with the type of equipment used to evaluate es by these various methods be highly developed prior to commencing runs for data.
- C. Additional work on the tracer gas method, as applied to reliability in use with spark-ignition engines, is indicated.
- D. The sampling check valve should be tested further, and its design changed, as tests indicate, to render it serviceable over long periods of operation.



INTRODUCTION AND GENERAL PROCEDURE

The importance of scavenging efficiency has increased in the past few years, with the increasing use of the 2 stroke engine. The difficulty presented is the lack of positive, accurate, simple means of determining the quantity for various R_s. Different methods have been devised in an effort to solve the problem. To the best of the knowledge of the authors, a chosen few of these methods have never been compared simultaneously, on the same engine, under the same operating conditions. This in substance, is what this thesis has attempted to do. Fig. I is a diagrammatic sketch of the entire equipment concerned with the four methods, and gives a general picture of how the work was done.

Since the "complete mixing" curve of es vs Rs was the only established reference curve, it will be used as a reference to permit relative evaluation of the results, and is included in all curves as such a reference.

The four methods chosen were (1) The Tracer Gas Method,

(2) The Gas Analysis of compression and expansion samples, (3)

The Gas Analysis of expansion samples only and (4) The I.S.A.C.

Method (Indicated Specific Air Consumption) referred to hereafter as Methods I thru IV respectively.

The Engine

A single cylinder 32 x 42" modified Waukesha, CFR two stroke, loop-scavenged, spark ignition engine was used. 100 octane



gasoline was injected.

Fig. XV shows the timing diagram of the porting and injection events.

Air inlet stream was carefully controlled by a standard orifice meter, regulator valves, and water Manometer. Inlet temperature was maintained thru Variac-controlled heater elements. Inlet temperature for all runs was 110°F.

Exhaust pressure was measured by manometer, on the exhaust tank, and ranged from .2 to .4 inches of mercury.

Spark advance was controlled by neon flash on a graduated dial.

Fuel was metered thru a rotameter to a Bosch injection pump.

with the decision to run at, or near best power conditions, it was necessary to obtain curves of best power information to permit setting the engine in a consistent manner. This was done by setting a scavenging ratio and speed on the engine, and reducing the fuel rate until the engine barely performed steadily. At the fuel setting, the spark advence was varied from 30° to 0°, and Brake load readings noted on the hydraulic scale manometer. The fuel was increased by an increment on the rotameter and the process repeated, and so on, until engine started missing due to excess fuel. For each increment on the rotameter, a plot of brake load versus spark advance gave a peak point of brake load. A plot of brake load "peaks" against fuel rate gave best power point fuel rate and spark advance. This process,



repeated 5 to 6 times for each speed, for varying R_s, mave best power curves with sufficient information to set the engine at, or near, best power for all conditions needed. These curves have been turned over to Sloam Laboratory for use with the engine.

Timed Sampling Valve

The valve used to draw samples for the gas analyses was the pressure element of a Cox Type VI Direct Pressure Indicator. Phasing of sample extractions was accomplished using the phasing equipment which is ordinarily part of the direct pressure indicator. The valve was solenoid-operated, by condenser discharge and opened against a spring. It started opening, and closed fully in 8 crank angle degrees at 1000 RFM, and in 11 crank angle degrees at 1400 RFM, according to tests run by Cox Co. Tests by Cox Co. also found that a pressure gauge in the sampling system read the pressure in the cylinder at the point in the cycle at which the valve completed its closing motion. A reasonable check of this value was obtained by comparison of values on the pressure gauge, in sampling system, with predicted pressures taken from indicator diagrams. The position of the valve in the engine head, and in the overall layout may be seen in Figs. I and XIV.

The Tracer Gas System

The flow of monomethylamine gas from the gas bottle into the inlet air stream was controlled by a water manometer across an orifice plate. Diameter of the orifice was .Ohl inches, and



meter readings and measuring inlet concentrations. Samples were drawn thru perforated collector tubes in the inlet and exhaust streams, and bubbled thru a solution of H2SO4, colored with 3 or 4 drops of medified methyl red indicator, thence thru a wet gas meter, to a surge tank, using a water aspirator. Since only one gas meter was available, inlet and exhaust samples were taken separately. Neutralization of the indicated H2SO4, by monomethylamine was signalled by a change in color of the bubbler solution from purple to clear water color. Further addition of methylamine turned solution bright green, indicating a basic solution. Titration burettes (2) completed the necessary equipment for determining the exact point at which neutralization occurred. A step-by-step procedure is included in Appendix A.

The Gas Analysis System

Samples at a chosen point in the engine cycle were passed from sampling system piping into a standard Orsat by displacing 100 cc of saturated salt solution. GO_2 , O_2 , and GO_3 , in the sample were removed in that order, by KOH, Oxorbent and Cosorbent (the latter two by Eurrell & Co.) solutions respectively. Four "passes" of the sample thru each of the three solutions were used.



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Figures II thru IX and the master curves of eg vs kg obtained from the data points shown on each sheet. The moints designated by on all curves indicate that their lack of precision is due to lack of technique, and all those points were taken prior to April 6, 1949. Points on curves of Nethods II and III designated by an indicate that their lack of precision is due to small leakage in the Cox sampling valve. This valve is constructed so that cylinder pressure tends to open it, against a spring force. The most minute failure of the lackage to noticeably affect results. Inexplained deviations from the preponderance of results are designated by on all curves.

The master curves are combined no speeds in Mas. . and XI, by method, in Mig. AII.

Table I is a numerical comparison of curve values of e_8 at resular increment of R_8 over the range from $R_8 \pm 1.0$ to $R_8 \pm 1.3$. From this table the comparison of absolute values, and percentage variation on the specified basis, can be seen at a slade, listed against speed and nothed. All curves exhibit a marked into of the rap between absolute values of e_8 at 1500 and 1000 RFM as as recreases from 1.80 to 1.00, for this particular entire. Sethods II and III show an intersection or joining of the two curves at $R_8 \pm 1.00$, while Lethods I and IV show an intersection



or joining at $R_s=1.00$. The slopes of the curves of all methods, for a particular speed, are very nearly the same at all points, and bear a marked resemblance to the slopes of the complete mixing curve at the same point. There seems to be no doubt that e_s is greater, for this engine, at 1400 RPM than at 1000 RPM, over the range $R_s=1.0$ to $R_s=1.8$.

Table II lists the authors' evaluations of and information on, the four mothods using a few chosen criteria that appear to be factors in choosing the method most suitable for use under various conditions. Table II in section on Conclusions.

Fig. XIII is the single curve of the results of the sampling check valve tests. The curve is judged to be good, as far as it goes. No opportunity to obtain estimates on precision, or to increase the number of points on the curve, was available.

Since three of the four methods depend on sampling valve performance on an expansion sample it follows that if the electrically timed sampler was replaced by the check valve sampler, the results would differ generally as the gas analysis, resulting from the performances of the two valves, differed. This difference in egrows Rg curves, would be direct relations between fuel-air ratios found by respective gas analyses, in Methods III and IV. The effect on Method II is more obscure. For purposes of illustration, a curve of egrows Rg, at 1000 RPM for Method IV has been modified by the ratio of (P/A) using timed sampler to (F/A) using sampling check valve, and placed on Fig. XIII for comparison with 1000 RPM Sampling Check Valve results.



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, variation =	hirh-low x 100	(3	٠. ٢٠٠	3.04	2.30	05° t	3.58	5.30	4.33	6.10		
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	Method	52.6	52.6	56.7	7.7 60 7.7	59.8	63.0	62.5	0-29	65.0	70.2	
	Method	4	51.3	76.	57.6	60.3	7.29.	53.57	2.99	66.0	9.69	
	Method	51.1	51.5	56.0	57.2	26.0	61.5	61.0	2-179	62.3	67.5	
	RPM	1000	14.00	1000	2400	0001	11,00	1000	00 [†] T	1000	11,00	
	B		7.00	6	1.20		7 01/- 7		20-1		000	



CONCLUSIONS

Table II summarizes the conclusions and evaluations of the four methods, with tabular information on various aspects of the problem for each method. This section will be devoted to enlarging on pertinent points in the table, and to the conclusions drawn regarding the sampling check valve.

Method I

evaluation of e_s is one of the moverning factors in accuracy, and must be done with extreme care. The location and type of sample collector used with the method is another factor of extreme importance in results obtained. The assumptions required are valid, and the figures for these correction factors as recommended in NACA Tech. Notes #838 are realistic. The erratic character of measurements of e_s are unexplained in some cases, and attributable to lack of technique in others. Control methods for the use of tracer gas are important, and must be of an accurate nature. The volume of the H₂SO_{||} used in bubblers, affected the result. Generally, the larger the volume of H₂SO_{||} the better the result. This factor must be balanced against the time consumed in neutralization of the H₂SO_{||} by the monomethylamine gas.

In its present condition, this method is limited in its scope, and development is required before it will be satisfactory



enough to be used, without reservation, on any engine, the exhaust stream of which, has a temperature of about 800°F, or higher. Indications are, that the acceptance of this method, in its present stage of development, as achieved in this work, would involve an individual problem of adaptation to a particular engine, and a means of comparing results to a dependable source, or calculations.

Method II

Method II is an unwieldy, cumbersome method in comparison with the other three. The extraction of two samples for each run, with an Orsat analysis of each is tedious work. The method is weakened immediately by the large number of assumptions required, although the results, which are good, do not reflect this weakness. One of the big disadvantages of the method is the large amount of time consumed in operation. The sensitivity of the sampling valve to leakage has a large effect on results obtained. Therefore, the preferred type of sampling valve is one which tends to close when exposed to cylinder pressure. The Orsat analyzer for flue gases appears to be completely satisfactory for this type of work.

In the face of results obtained using Methods I, III, and IV, Method II is adjudged least desirable of all methods.

Method III

As the preferred method, this procedure has as a sole disadvantage, the necessity for dependence on the performance of



the sampling valve, the dependability of the gas analysis charts, and the extraction of a representative sample; its advantages outweigh this, however, since it gives precise, accurate, quick results. Calculations are direct and simple. A good sampling valve, properly located, would provide the answer to any weakness in the method. The sampling check valve tested in this work appears to be a step in this direction. Very satisfactory results were obtained.

Method IV

The curves of es vs Rs obtained using this method were the most accurate of all results. No check runs were required or made, hence no information on duplication of results is available. With necessary equipment available, i.e. a 4 stroke engine "comparable" to the 2 stroke being examined, and indicating equipment, the method is desirable of use. The procedure is unwieldy, tedious, and time-consuming. Generally speaking, the method is very satisfactory on a basis of results obtained, but practical requirements limit the availability and desirability of the procedure.

Sampling Check Valve

Since only five hours operation were recorded on this valve, no positive statement of its worth can be made. The operation of the valve was sufficiently satisfactory, during the five-hour period to permit a positive conclusion to be drawn, i.e. that further work with this valve, or a similar design is



definitely indicated. The test was terminated because the valve cloggel, and ceased to operate. Modifications of design could alleviate this condition. Fig. XIII aids the contention that the valve is worthy of further work. The ultimate elimination of the ordinary complicated, bulky, timed sampler, in favor of the small simple, flexible check sampler, for use with Method III appears most favorable.



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Calculations (single point on es vs Ks curvo)	Extremely quick and simple. Can be done in three to five minutes once equations have been adapted to use.	Most complicated and longest calculation required, of all four methods. Requires 20 minutes of calculation during run may be made.	Calculations occupy no more than a min- ute, and can he per- formed in one operation of slide rule.	Aequires about one hour to planimeter indicator cards, and utilize data to evaluate es.
Difficulty of Setup	20 man-hours of labor plus nurchase time. Considerable pipefitting and reneral machine work is necessary.	Phasing equip- ment and sam- pler used in this work could be set up in 1 man-hour by anyone familiar with the equip- ment.	As in Method	About 15 min- utes sotup time on H.I.T. Indi- cator, and 30 minutes setup and warmup time for each engine.
Egulpment Reeded	(2.24 ft3) containing 40 lbs. monomethylamine under vapor pressure. (About 30 psig at room temps.) 2) l Wet gas moter. 3) Titration and bubbling glassware. bling glassware. 4) Miscellaneous glass and copper tubing.	1) Sampling valve. 2) Phasing accessories. 3) Orsat equipment. 4) Accurate pressure gauge, 0-200 lbs. psi.	As above, for Method II.	1) 4-stroke engine of comparable bore and stroke with 2 stroke being examined. 2) Pressure indicating equipment.
Time Consumed to Obtain Point on es vs Rs curve	25 winutes to obtain and titrate an exhaust and inlet sample and carry thru calculations.	l hr. and 30 min. to obtain and analyze a com- pression and ex- pansion sample, and carry thmu calculations.	25 minutes to ob- tain and analyze an expansion sample and carry thru calculations.	l hr., 15 min. to obtain indicator cards, fuel and air data on each of 4 stroke and 2 stroke engines, in succession.
Method	H	1-4 1-4	H H H	AT



Assumption, wither Required or Desirable

Method

1

1) Assumption required as to how much of monomethylamine in air retained in cylinder dissociates during combustion. 96% was used in this work.

2) A correction for the volume change of fuel air mixture, due to combustion (shrinkage factor). 7.5% was used in

this work.

1) That sampler is taking a representative sample.

2) Temporature of exhaust gases in cycle is equal to temperatures calculated for

fuel air cycle.

3) Temperature loss of exhaust during "blowdown" = 300°F.

4) Isentropic expansion and compression during late expansion and early compression strokes.

5) That actual F/A is that determined by analysis of residual gas samples.

1) That sampling valve is taking a representative sample.

2) That actual F/A is that determined by analysis of expansion samples.

1) That a 4 stroke engine, of comparable bore and stroke to a 2 stroke, if run at the same piston speed, outlet jacket water temperature, inlet air temperature, and compression ratio as the 2 stroke, will exhibit the same combustion characteristics in the cylinder as the 2-stroke. This is assumed for injection or pre-mixed charge operation, or for any combination thereof.

Consistency of Regults Variations due to "experiment error": This method was the most difficult to control and showed the greatest variation in eg when an attempt was made to duplicate results. These variations of eg were as high as .10, but successive runs for duplication tended toward avoraging to a good result - Details in Appendix A.

Method II showed the second larrest variations of es when duplication of results was attempted. Technique played a large part in this method, along with a tondercy to leakage exhibited by the sampling valve after from 4-3 hours of operation, and it is felt that these two factors account for a high percentage of errors.

Method III, with Method IV shared the position of showing little or no deviation from former results when duplication was attempted.

The absolute values of es obtained by this method were close to an average of all four methods and all points obtained lay on, or very nearly on a shooth curve of es vs Rs.



Method

General Remarks

and 1.80, based on results of other methods, and inaccurate at Rs = 1.0 region Rs = 1.2 to Rs = 1.65 results were acceptable. It is considered to be a potentially suitable method for the purpose of measuring es, but must be more effectively adapted to the spark - ignition engine. Were this Method I results were the most difficult to evaluate in terms of a smooth adaptation to be accomplished nore completely than in this work, with attendant reliability, this method would be judged as desirable, but inferior to Method III.

but the cumbersome procedure, long calculations, time consumed, technique The results are good, requirement, and the numerous assumptions required, are all strong de-Method II is classed as the most unsatisfactory. tractors from a choice of this method.

H

sistency of performance and accuracy are concerned. Method III is chosen as a method preferred, because it does not require a 4 atroke engine, of certain size, nor does it require indicating equipment. Despite the fact that this method requires an expensive sampling valve and an Orsat, it is Methods III and IV would be classed as equally effective as far as conclassified as the preferred of the four methods, on an overall basis. Using as criteria, precision, and consistency, this method ranks with III as clearly superior to I and II. The two big drawbacks connected with necessity for indicator equipment. The paperwork involved per point on this method are the necessity of having available a 4 stroke engine of a size within arbitrary limits, rigged for laboratory work, and the es vs Rs curve is fairly cumbersome.



RECOMMENDATIONS

The assumption that perpetuation of this type of work is desirable indicates three paths along which efforts should be directed. The first of these is the development of a method for measuring e_s, which, in general, would consist of injection, in known amounts, into the cylinder of a 2 stroke engine, after the ports have closed, an element or compound which could be identified quantitatively in the exhaust stream. Obvious requirements for such a material, that present themselves are:

(1) It must not be present in the products of combustion or in the fresh charge in any quantity (2) It must maintain its identity under combustion temperatures (3) Accurate measurement of its presence in the exhaust stream must be possible and facile, and (4) It must lend itself to accurate injection against about 600 psi.

The problem outlined above is one of a chemical and practical nature. Helium suggests itself as a choice for items (1) and (2), above. No further examination of the problem will be made.

The second subject for further work is the tracer gas method. The results obtained between the end points on the es vs Rs curve are quite acceptable. The poor results obtained consistently at the end points are unexplained. There is room for improvement and variation in (1) The type of sample collector



tubing used (2) Its location, and (3) Control methods used.

The ideas behind the method are sound, and worth an effort in the direction of practical improvement.

The further development of the sampling check valve is the third direction in which more investigation is considered worthy. The only fault to be found with the sampling check valve used is its tendency to clog with foreign matter, and cease operating. In order to enter the valve a single piece of foreign matter had to be less than 1/16" in any dimension. Once inside the valve however, the flow thru the valve seat was split four ways thru apertures 1/2 the diameter of the entrance. Thus the particle which passed the entrance orifice, may or may not have continued along the line. If further testing of this valve points up the failure due to clogging, it is recommended that a design be used that will make the entrance orifice of the valve the smallest restriction in the system from cylinder to Orsat equipment. The size of the valve used in this work was restricted only by the size of the exhaust port of the engine.

As a general recommendation, it is felt that thorough familiarity with technique involved in these methods should be cultivated by practice runs, before record runs are made.



APPENDIX A

Detailed Procedure and Sample Calculations

This section will be devoted to a short description of the engine used, and its peculiarities, followed by a step-by-step detailed description of each method in succession. A typical data run will be chosen, the engine conditions specified, and the comments on each method will be accompanied, or followed by, a set of sample calculations.

The Engine

Operation of the engine was found to be unsatisfactory for the purposes of this work when a single spark plug located near the center of the head, was used. Accordingly, an additional ignition system to a side plug was added, and performance from then on was suitable. Injection took place from a nozzle placed near the center of the head. The sampling valve was placed symmetrically with the additional side spark plug. Details of the location of plugs, injector and sampling valves may be seen in Fig. XIV.

Excellent control of air inlet temperature was obtained by mounting a "Variac" in series with the heater element in inlet piping. Inlet air temp was measured by mercury thermometer in inlet air mixing tank. Exhaust stream pressure was recorded by mercury manometer on the exhaust cooling and surge tank. See Fig. XV for diagram of port and injection timing.



Spark plugs used were 3G Aviation Spark Plugs #157, and gave steady firing, with the dual ignition system. It is to be noted that while the engine ran fairly steadily at optimum settings for a certain condition of R_s and speed, using the head plug alone, any departure from optimum settings caused prohibitive misfiring, and under no conditions could anything but spasmodic firing be obtained using the side plug alone. The simultaneous use of the head and side plugs resulted in steady firing over a reasonably large variation from optimum settings. Fuel rate was controlled by rotameter between supply and injection pump.

Engine Set-Up

For the purposes of sample calculations, a run made at R_s = 1.4 and 1400 RPM will be used in conjunction with Method I, II, and III.

Engine Conditions

$$R_{S} = \frac{\text{Air delivery rate}}{\rho_{S} \text{ N V}_{D} \frac{r}{r-1}} = \frac{M_{\text{adel.}}}{\rho_{S} \text{ N V}_{D} \frac{r}{r-1}}$$

$$\rho_s = .0765 \times P_{\frac{900}{2992}} \times \frac{520}{T_1}, \quad V_0 = 37.33 \text{ ln}^3, \quad \frac{r}{r-1} = \frac{6.72}{5.72}$$



$$V_{\rm D} = \frac{r}{r-1} = .03375 = \frac{N \cdot p_{\rm O}}{T_{\rm i}} = \frac{1 \cdot s \cdot sir}{rir}$$
, $N = rpm$

$$p_{\rm O} = 1 \cdot n \cdot c \cdot h_{\rm C}$$

For these conditions
$$s: V_D = 2.43 \frac{lbs air}{min}$$
, $T_i = 370^{2} n$

For
$$R_s = 1.4$$
, $R_{del} = 1.4 \times 2.40 = 3.47 \frac{1bs air}{min}$

Air flow was set by standard orifice meter, resulators, and water manometer. Fuel rate and spark advance were taken from best power curves as determined previously. Fuel rate = .00223 lbs fuel/sec. Spark advance = 15.5°. This completed the setup of the engine for a particular run.

Method I - Reference MaCA Tech Notes #735, by Schweitzer and DeLuca. See Fty. I for general setup.

Symbols

Monomethyla line CH3NH2

Sulphuric Acid H2SO4

Sodlum Hydroxide NaCH

Ex. (.1N)H2SO4 = .1 normality solution of H2SO4

W = weight of CH3NH2 in lbs.

 $^{\rm IN}$ H₂SO_{l1} = molecular wt. H₂SO_{l1} = 98

mcH3NH2 = molecular wt. CH3NH2 = 31



X_{cc} = Volume of (_N)H₂SO), used in sample in c.c.

Accorr = Xcc corrected to end point of neutralization

Yec = Vol. of NaOH used to find end point in c.c.

 V_{c} = volume passed thru gasmeter during a sample extraction - ft³

Tg = Temperature of sample passing thru meter - OR

P = Pressure of sample passing thru meter - P = 1
P atmosphere

The basis on which the method is used successfully is that CH_3NH_2 , if bubbled through a solution of H_2SO_4 will tend to neutralize it. If the H_2SO_4 solution is colored by an organic indicator, the passage of the solution from acid to neutral to base will be signalled by a change from, in this case, purple to clear water color to green. An equation may be set up:

Mols = Mols

At neutralization: Weight of H₂SO₄ in sample solution Mol. wt. of H₂SO₄

Equivalent weight of H2SO4 = 149

$$\frac{49 \times 10^{-3} \, (\text{N}) \, \text{H}_2 \, \text{SO}_{\frac{1}{4}}}{98} = \frac{\text{W}}{31} \, , \quad (\text{N}) \, \text{in gm/cc}}{(.112 \text{N}) \, \text{H}_2 \, \text{SO}_{\frac{1}{4}}} \text{ was used.}$$



If Xcc is volume of H2SO4 solution sample of normality (_1),

$$W = (31 \times \frac{h_0}{98} \times 10^{-3})(h)_{\text{H}_2\text{SO}_h} (X_{\text{cc}_{\text{corr}}}) \times \frac{1}{h5h} = 1\text{bs}.$$

Concentration in inlet air =
$$\frac{W}{H_{air}}$$
 = $\frac{3.32 \cdot 10^{-5} h_{eccorr}}{V_{7} \cdot .0765 \cdot F_{2}}$ = $\frac{520}{29.92 \cdot 13.5}$

Monomethylamine is metered into the inlet air stream by use of a manometer across an orifice (The .041" orifice dror was found to be from 6 to 45 inches No.0 for satisfactory concentrations). This gas mixes with the inlet air stream, and passes into the engine. Some passes thru the engine, in bypassed air, and some dissociates in the combustion process within the cylinder. Samples are drawn thru bubblers and the gas meter, at a point in inlet and exhaust streams. The tracer gas in these samples neutralizes the acid in the bubblers. The end point of neutralization was calculated as follows: 10cc of (.112N)H2SO1 in distilled water was placed in the bubbler. The air + CH3NH2 sample was drawn from the inlet stream thru the bubbler for about 10 mins, until the purple solution neutralized and turned green, indicating base. The sample is thus taken. 4 cc of (.112N)H2Solp was added by burette and neutralized with Yoc of (.109N) NaOH (Solution is clear water color) also from a burette. From this,



$$X_{cc}$$
 - X_{cc} - $\frac{.109}{.112}$ Y_{cc}

For this run, Xcc = 10 + 4.1 Ycc = 4.1

..
$$X_{cc_{corr}} = 14.1 - (.974)(4.1) = 10.1 cc for inlet sample

 $T_g = 73^{\circ}F = 533^{\circ}R$
 $V_g = .37 \text{ ft}^3$$$

Inlet concentration = (.964) • (10.1) • (.533) • (10⁻⁶)

The procedure was repeated for the exhaust sample, using 10 cc (.112N)H2SO1. When solution turned green, sample was completed and end point determined in the same manner as for inlet sample. Exhaust sample takes about 15 min.

Results were
$$X_{cc} = 11.0$$
 $Y_{cc} = 3.3$ $X_{cc} = 10.8$ $V_{g} = .650 \text{ ft}^3$ $T_{g} = .74^{\circ}\text{F} = .534^{\circ}\text{R}$

At this point, the concentration of the exhaust sample could be found, as in inlet sample, but this is not necessary. Concentration of inlet samples were calculated as a means of continuous calibration of the tracer gas orifice. Except for small effects, known as "burning efficiency," and "shrinkage factor," to be discussed further along, it can easily be shown that the ratio of the concentrations in inlet and exhaust samples is also the ratio of "air delivered" to "air not retained "in cylinder. Thus, if 3 lbs of air flows into engine carrying .1 lbs of tracer



exhaust stream passes 3 lbs of air and residuals, but 2/3 of the monomethylamine is lost in combustion, hence the concentration of the exhaust stream is .0333 as opposed to .100 in the

inlet stream.
$$\frac{.0333}{\frac{100}{2}} = \frac{1}{3}$$

If α = ratio of exhaust concentration to inlet concentration From (1), exhaust concentration = ∞ = $\left\{\begin{array}{c} \cdot 964 & \text{Xcc}_{corr} \cdot \text{T}_{S} \\ \hline \text{inlet concentration} \end{array}\right\}$ exhaust

$$\mathcal{C} = \frac{(X_{\text{cc}}^{\text{corr}^{\text{T}}g)}_{\text{exh}}}{(X_{\text{cc}}^{\text{corr}^{\text{T}}g)}_{\text{inlet}}} \cdot \frac{(V_g)_{\text{inlet}}}{(V_g)_{\text{exhaust}}}$$
(2)

Also - Air not retained - Air delivered - Air retained
Air delivered Air delivered

... of = 1 -
$$\frac{\text{Air retained}}{\text{Air delivered}}$$
 = 1 - Γ where Γ = retention ratio Γ = $\frac{e_s}{R_s}$

$$\therefore \Gamma = 1 - \infty \tag{3}$$

This would be the solution of the problem, since we know the air delivered to the engine, were it not for the fact that despite the high combustion temperatures, all the tracer gas, in the cylinder, in the air retained, does not dissociate during



combustion. The correction factor is known as burning efficiency, eb. NACA Tech. Notes #838 recommend a value of .96 for eb, and this figure was used. In addition, the volumes of the inlet and exhaust streams are not equal due to the combustion process, the correction for which was introduced as a shrinkage factor, s. NACA Tech Notes #838 gave curves of "s" versus F/A for Diesel fuels. These curves were extrapolated and a value of .075 was assumed for s. eb and s remained the same for all runs. Introducing the two correction factors into (3) gives

$$\Gamma = \frac{1 - \alpha}{e_b - s\alpha} \qquad (!_{\downarrow})$$

The detailed derivation of this expression may be found in NACA Tech Notes #838.

For the run being examined, by using (2)

$$\alpha = \frac{({^{X}_{cc}}_{corr} \cdot {^{T}_{g}})_{exh}}{({^{X}_{cc}}_{corr} \cdot {^{T}_{g}})_{inlet}} \cdot \frac{({^{V}_{g}})_{inlet}}{({^{V}_{g}})_{exhaust}} = \frac{10.8}{10.1} \times \frac{534}{533} \times \frac{.37}{.65}$$

$$\alpha = \frac{.61}{...61}$$

From (4)

$$\Gamma = \frac{1 - .61}{.96 - (.075)(61)} = .427$$

Air delivered = 3.47 lbs/min. from engine setup.

... Air retained = 3.47 · .427 = 1.481 1bs/min

$$e_s = \frac{\text{Air retained}}{\rho_s N V_D \frac{r}{2.480}} = \frac{.60}{2.480}$$



There are several comments to be made on this method of determining es.

Monomethylamine, as an "educated" tracer gas, dissociating in the combustion process, but not in the exhaust stream, rives a reasonably satisfactory account of itself. However, certain precautions must be taken, in use with a spark implicant wo stroke engine. The gas combines very readily with water, hence if condensation occurs anywhere in the system thru which the gas passes, the condensation will pick up the monomethylamine, removing it both from the system and from the calculations. In addition, there seems to be no accurate information on the temperature at which CH2MH2 dissociates. Schweitzer and DeLuca, in the reference NACA publication, imply that at any temperature above 800°F, it can be expected that the gas will dissociate. Since the residual gases of the exhaust are at about 2000°F, and comprise more than half the exhaust stream, the temperature of the exhaust stream is more than likely in the vicinity of 1500°F. It follows that, unless the exhaust stream sample can be taken very close to exhaust ports, and removed from the high terporature area rapidly, the tracer cas will dissociate outside the cylinder, with accompanying loss of accuracy in measurement of eg. In the setup, the stream passed from the ports thru a foot long duct to a water-cooled surge tank, thence to the laboratory trench. The first setup for tracer was work took inlet samples about 2 inches before entry of stream into ports,



and erhaust samples on the downstream side of the exhaust surge tank, as outlined in NACA Tech kotes #832. Thoroughness of ining of inlat air stream and monomethylamine was assumed, based on the following: Tracer gas was injected into inlat air stream in a mixing tank, the mixture passing from the tank, through three 90° elbows into a second mixing tank, thence to inlet ports. Results were poor, and duplication of results was impossible. The MACA reference had stressed the necessity for taking several samples at intervals along the exhaust line. In the face of results, it was decided to move the sampling post to the duct between exhaust port and surge tank, and take only one exhaust sample, just outside the ports. This would eliminate possibilities of ourning in exhaust stream, and the ressibility of loss of tracer gas to confensation in the cooled our o tank. Accordingly, this was done using the same & inch diameter perforated steel tube collector, placed across the duct. Results were equally poor, as in first tests, and it was reasoned that the collector, by virtue of its diameter being some three times the diameter of the smallest tubing in the sampling system, was permitting the tracer gas ultimately intended for the bubbler, to slow down or stop in the collector, which itself was directly in the residual gas blast. The slowdown permitted dissociation of monomethylamine in the high temperature area. In an effort to shorten the time in which the monomethylamine was exposed to high temperature, higher



velocity of gases thru the exhaust duct to sampling bubbler was schieved by using a single tube, of inside diameter 3/64" placed in the blast of center exhaust ports, an' the maximum speed of bubbling thru the H2SOh solution was used. The absolute values of eg, obtained at this point dropped about 30% into the region occupied by the es values by other methods. Duplication of results, however, was still unreliable. A 5/16" inside diameter tube, used in a similar manner as the 3/64" single tube yielded the same characteristics. The final collector was a straight tube, placed across the duct, of 5/15" inside diameter, fed by 4 smaller tubes, mounted in the direction of flow, their apertures mounted at the centerline of each of the 4 exhaust ports, at a distance of about 1.5 inches from the ports. Details may be seen in Fig. . This type of collector gave values of es which compared closely with values obtained by the other methods could be duplicated within .0%, and fell on a reasonably smooth curve of es vs Rs, except at the extreme ends of the curve, where Rs = 1.0 and Rs = 1.8. Repeated check runs at these two points gave unexplained results, of es about .04 high at Rs = 1.0 and of es about .05 low at Rs = 1.6. The terms "high" and "low" refer to the smooth curve drawn through values of es in between the end values.



APPENDIX A (cont'd)

Lethod II - Cas Analysis of Compression and Exhaust

Samples

Symbols

Mr = mass of residuals in cylinder - lbs.

Mm = mass of fresh mixture in cylinder - lbs. (Fresh air only.)

Mc = Total mass of gas in cylinder - lbs.

mr = molecular weight of residual gas - lbs.

lb nol

mm = molecular weight of fresh mixture - lbs/lb mol.

mc = molecular weight of cylinder gas - lbs/1b mol.

 χ = 02 fraction by volume in $M_{\rm m}$ - dimensionless.

y = 02 fraction by volume in Mr - dimensionless.

 $Z = 0_2$ fraction by volume in M_c - dimensionless.

R = universal gas constant = 1544 ft. 1bs

 $R = \frac{R}{\text{mol. weight}} = \frac{1b \text{ mols. ft}}{\text{oR}}$

Subscripts

r - residuals o - point at which compression sample taken

m - fresh charge s - point at which expansion sample taken

c - cylinder contents e - exhaust

i - inlet d - displacement

a - air l - indicates beginning of compression stroke on indicator diagram

Other symbols, with appropriate subscripts as previously defined.



Development of Equations

An equation may be set up: Mols of Oxygen = Mols of Oxygen

$$\left[x \cdot \left[\frac{M_{m}}{m_{m}}\right] + y \left[\frac{M_{r}}{m_{r}}\right] = z \left[\frac{M_{c}}{m_{c}}\right] = z \left[\frac{M_{m}}{m_{m}} + \frac{M_{r}}{m_{r}}\right]$$

Prom this --
$$\frac{M_r}{M_m} = \frac{x-z}{z-y} \cdot \frac{m_r}{m_m}$$
 (1)

$$\frac{M_{\rm c}}{M_{\rm m}} = \frac{1 + \frac{M_{\rm r}}{M_{\rm m}}}{\frac{M_{\rm c}}{M_{\rm c}}} = \frac{1}{1 + \frac{M_{\rm r}}{M_{\rm m}}} \tag{2}$$

$$\frac{M_{r}}{M_{c}} + \frac{M_{m}}{M_{c}} = 1 \qquad \frac{M_{r}}{M_{c}} = 1 - \frac{M_{m}}{M_{c}}$$
 (3)

At point "O" where compression sample is taken,

$$p_{o}V_{o} = M_{c}RT_{o} \qquad R = \frac{R}{m_{c}}$$

$$M_{c} = \frac{p_{o}V_{o}}{R_{c}T_{o}} = \frac{p_{o}V_{o}}{R} \frac{m_{c}}{T_{o}}$$
(14)

$$e_{s} = \frac{M_{m}}{\rho s^{V} \frac{r}{r-1}} \qquad \text{(one cycle)} \qquad \rho_{s} \text{ at } p_{e}, T_{1}$$

$$\rho_s = \frac{1}{v_s} \qquad p_\theta \cdot \frac{1}{\rho_s} = R_m T_1 \qquad R_m = \frac{\overline{R}}{m_m}$$

$$\rho s = \frac{p_{e}m_{m}}{\overline{R}T_{1}}$$



Combine with (2),
$$e_s = \frac{M_c}{1 + \frac{M_r}{M_m}} \cdot \frac{RT_i}{p_e \cdot m_m \cdot V_c}$$

Combine with (4),
$$e_s = \frac{1}{1 + \frac{M_r}{M_m}} \cdot \frac{p_o V_{om_c}}{\vec{\chi} T_o} \cdot \frac{\vec{\chi} T_i}{p_e m_m \cdot V_c}$$

$$e_{s} = \frac{p_{o}}{p_{e}} \cdot \frac{V_{o}}{V_{c}} \cdot \frac{m_{c}}{m_{m}} \cdot \frac{T_{i}}{T_{o}} \cdot \frac{1}{1 + \frac{M_{r}}{M_{m}}}$$
 (5)

All quantities in (5) are known, or can be found directly, except m_c , T_o , and m_r . The latter is concealed in M_r as shown in (1).

po - Taken from gauge in sampling system - psia

pe - Exhaust tank manometer applied to atmospheric pressure - psia

Vo - Taken from knowledge of point at which compression sample is taken. See Fig. XVIII.

Vc - Engine dimensions, compression ratio

mm - Taken as molecular weight of air

Ti - Taken from inlet tank thermometer

The evaluation of the three unknowns was accomplished as follows, in the order given.

1) mr - The results of the analysis of the expansion sample



were used to enter two charts of exhaust was analyses of 4-stroke spark-ignition engines. Specific reference to charts may be found in Appendix D - Bibliography.

They are reproduced in Fig. XVI.

The charts were plots of elements and compounds as ordinates in percentage by volume, against fuel-air ratio.

The analysis of the expansion sample gave points on the CO, and CO2 curves in both charts, and permitted the establishment of a vertical line, representing a particular fuel-air ratio.

See Comments, on this method, for information on oxygen in analysis. The line on each chart allowed a buildup of the relecular weight of residuals, by fractions of each component in exhaust gases. The desired fractions were indicated by the intersection of the curve of the particular component and the established vertical line. Thus 70% nitrogen contributed

.70 · 28 = 19.60 lbs to m_r, 8% CO2 contributed .03 · M4 = 3.52 lbs., and so on. The results of these two processes, one on each chart were averaged, and this was taken as m_r. From m_r, x, y, z, and

m_m, M_r could be calculated, followed by M_m, and M_r.

M_m

2)
$$\frac{m_{c}}{m_{c}} = m_{r}M_{r} + m_{r}M_{m}$$

$$m_{c} = m_{r} \frac{M_{r}}{M_{c}} + m_{m} \frac{M_{m}}{M_{c}}$$
(6)

All quantities necessary to determine mc, are known.



3) T_0 - The expansion sample is taken at a point "\$' just prior to opening of exhaust ports. I good approximation of T_p at this point may be taken from Pig. TVII by entering with the % theoretical fuel (determined by dividing the everage of the fuel-air ratios taken from charts Figure XVI, by .0665), and from the proper expansion ratio line, reading the temperature of residuals in $^{\circ}$ F absolute. The expansion ratio in this case is the volume above the piston when exhaust ports just open, divided by the clearance volume.

Assume isentropic expansion from ps to pe, k = 1.35

$$T_r'$$
 at pressure $p_e = T_r$ (chart) $\left[\frac{p_e}{p_s}\right]^{\left(\frac{k-1}{k}\right)}$

Assume 300°F loss in temperature during "blowdown" process, and the temperature of the residual cases at the beginning of compression, $T_g = T_r \cdot \left(\frac{p_e}{p_s}\right)^{\left(\frac{k-1}{k}\right)} - 300$

ps may be taken from the gauge in sampling system in psia.

At this point, mixture of scavenging air and residuals has taken place, and from an assumption that mixing takes place adiabatically at constant pressure:

 $M_{c}h_{1} = M_{m}h_{1} + M_{r}h_{r}$, where h is unit enthalpy. For perfect gases, and assuming c_{p} is the tame for M_{c} , M_{m} , and M_{r} , h is a function of T alone, hence $T_{1} = \frac{M_{m}}{M_{c}}$ $T_{1} + \frac{M_{r}}{M_{c}}$ T_{g}



assume isentropic compression from beginning of compression to noint "O" at which compression sample was taken.

$$T_0 = T_1 \left(\frac{p_0}{p_e} \right)^{\left(\frac{k-1}{k}\right)}$$

es may now be evaluated from (5).

Comments on Method II

While the setup work on this method is small, the method itself is cumbersome, and, in light of the results of the other three methods, the least desirable of all methods used. The precision is dependent almost entirely on technique, and type and location of the sampling valve. Technique can be covered be a recommendation that several "dry" runs be made prior to taking data for record. The operation of the Orsat equipment is simple, but consistency of results demands practice.

In this work the Cox sampling valve was located in a position shown in detail in Fig. XIV, and remained in that position for all data taken. This location was chosen arbitrarily, and was influenced by the holes already in the engine head and side of the head, and the fact that the injector and one plug had to remain in the head, for proper operation of the engine. All the was analyses of expansion samples showed from 3 to 7,5 oxygen by volume when it was known that the engine was running "rich," and that the correct oxygen percentage should have been about 0.6%. The CO, and CO2 contents of samples indicated a consistent fuel-air ratio, taken from the was analysis charts, of from



.082 to .089, and previous work had established that the enrine was at, or near, best-power conditions. The discrepancy in the O2 values obtained by analysis was attributed to poor mixing in the location of the sampling valve, and 02 results of analysis were ignored. When the fuel-air ratio was established on the analysis charts, using the CO and CO2 fractions found by analysis, the Op fraction used in calculations was that shown on the charts. For all runs, the values of 02 fraction on the charts, at the established F/A, were .3% for D'Alleva & Lovell's charts, and 1.0% for Gerrish and Meems! chart. These were averaged to .6%, and for all data runs, the quantity "y," the 02 fraction by volume, was taken as .6%. Disrecarding the 02 fraction by Orsat analyses was an arbitrary movo, dictated by various circumstances, of a practical nature. First, the "poor mixing" theory was aided by the fact that an additional spark plus mounted in the side of the head, in a position opposite the sampling valve, but symmetrical to it, with respect to inlet and exhaust ports, was incapable of running the engine alone, indicating that that area was one of incomplete mixing. Second, time was judged to be sufficiently closely scheduled as to preclude relocation of valve and the attendant machine work. Third, since the only suitable location would have been in the head, rather than the side of the head, and the head was taken up by a necessary plug and injector, the sampling valve remained in its initial location. Fourth, and most practical, the results were good. The presence



of the relatively large amounts of free 02 affected the CO, and CO2 fractions, causing a horizontal "spread" on each cas analysis chart. However, splitting the "spread" with a vertical line, cave fuel-air ratios, the values of which checked almost identically between charts, and contributed to satisfactory results in evaluating es.

A preferable situation would have been sufficient time and space to devote to trials of positions for the sampling valve, to obviate the procedure used in this work, and for future work that may be done, it is recommended that such arrangements be made.

The CO, and CO₂ fractions found by analysis were, generally speaking, further apart along the abscissa axis, on the charts, when the O₂ content of the gas analysis was high, and closer together when the O₂ fraction was low, indicating that better mixing should lead to better definition of fuel-air ratio on the charts.

Sample calculations

Using the same engine set-up as for Method I:

The exhaust sample was taken at 111° crank angle - point "s."

Fig XX shows position of points "o" and "s" on a typical indicator card. The values of "o" and "s" in crank angle degrees was the same for all runs. Gauge pressure, 46 psi , ps = 46 + 14.5 = 60.5 psia. hesults of Orsat analysis: CO2 = 7.6%; CO, 6.0%; O2, 4.4%.



Using CO₂ and CO fractions to enter charts (see Fig. XVI-0), F/A = .0375 on both charts. A vertical line is drawn on each chart at F/A = .0375. The buildup of molecular weight of residuals from charts:

Find mr - Gerrish and Meems Chart and D'Alleva and Lovell Chart

Component	Nol. Wt.	% by G & M	Volume D'& L	Weight G & M D & L	
N ₂	28	78.0	69.0	21.85	19.3
co2	44	4.8	8.0	3.69	3.52
CO	2 8	7.6	7.2	2.12	2.01
^{II} 2	2	3.9	3.0	.08	.06
02	32	1.0	•3	•32	.10
CH	16	.6	•5	.09	.08
H ₂ 0	18	W0 W0	13.0	the tity	2.34
		99.5	101.0	28.15	27.39

$$... M_r = \frac{28.15 + 27.39}{2} = \frac{27.77}{2} \text{ lbs/lb mol}$$

 0_2 fraction = .6% by volume = y

The compression sample was taken at point "o" at 317° crank angle. sauce pressure 51; psi, $p_{\circ} = \underline{68.5}$ psia.

$$V_o$$
 from Fig. XVIII = 12.3 in 3 $V_c = 13.89$ in 3

Analysis results $CO_2 = .6$, and $O_2 = 19.0\% = 8$

 $m_{\rm m} = 28.93$ (mol. wt. of air) and 0_2 in sir = 21.2 = x



$$\frac{M_{P}}{M_{H}} = \frac{x-z}{z-y} \cdot \frac{M_{P}}{M_{H}} = \frac{21-19.0}{19.0-.6} \cdot \frac{27.77}{28.93} = \frac{.104}{...}$$

$$\frac{N_{\rm m}}{N_{\rm c}} = \frac{1}{1 + \frac{M_{\rm r}}{M_{\rm m}}} = \frac{1}{1.104} = \frac{.905}{.905}$$

$$\frac{M_r}{M_c} = 1 - \frac{M_m}{M_c} = \frac{.095}{.000}$$

$$m_c = m_m \frac{M_m}{M_c} + m_r \frac{M_r}{M_c} = (28.93)(.905) + (27.77)(.095) = \frac{28.79}{100} \frac{100}{100}$$

Tr from Pig. XVII, at expansion ratio of 5.20 is 3170°R

$$T_{R} = T_{r} \left[\frac{p_{e}}{p_{s}} \right]^{\left(\frac{k-1}{k}\right)} - \frac{k}{300^{\circ}}$$

$$= 3170^{\circ} \left[\frac{11 \cdot 7}{60 \cdot 5} \right]^{\left(\cdot 26\right)} - 300^{\circ} = 2200 - 300 = \frac{1900^{\circ}R}{1}$$

$$T_{1} = \frac{M_{m}}{M_{c}} T_{1} + \frac{M_{r}}{M_{c}} T_{s}$$

$$= (1905)(570) + (1905)(1900) = 607^{\circ}R$$

$$= (.905)(570) + (.095)(1900) = 697$$
°R

$$T_0 = T_1 \left[\frac{p_0}{p_e} \right]^{\left(\frac{k-1}{k} \right)} = 697 \left[\frac{68.5}{14.7} \right]^{\left(.26 \right)} = 10100$$
R

Solve for es

$$e_{s} = \frac{p_{o}}{p_{e}} \cdot \frac{V_{o}}{V_{c}} \cdot \frac{m_{c}}{m_{r_{i}}} \cdot \frac{T_{i}}{T_{o}} \cdot \frac{1}{1 + \frac{M_{r}}{M_{r_{i}}}}$$

$$= \frac{68.5}{14.7} \cdot \frac{12.3}{13.69} \cdot \frac{28.79}{28.93} \cdot \frac{570}{1040} \cdot \frac{1}{1.104} = .645$$



APPLADRX A (cont'd)

Method III - Gas malysis of Expansion Cample Alone

The symbols and subscripts used in this method are the same as those used in Method II.

The procedure in this method is identical with that of Method II up to, and including the establishment of a vertical line on the Gerrish & Meems, and D'Alleva & Lovell charts of exhaust gas analysis. When the fuel-air ratio has been found from this line, the method procedure branches off from Method II, as follows:

F/A = lbs fuel, based on air retained in the cylinder lbs air

(Injection). This expression is true for a single cycle, or a rate of flow.

and lbs air/min =
$$\frac{\text{lbs fuel/min}}{\text{F/A}} = \frac{\text{Fuel rate}}{\text{F/A}}$$

Fuel rate and F/A are known, hence Ma can be found. retained

$$e_8 = \frac{M_{a_{ret}}}{\rho s^{NV}c}$$

Sample Calculations

Using the same engine setup as for Method I;

 $\rho_{s}^{NV_{c}} = 2.48 \text{ lbs/min}$ F/A from analysis = .0875 Fuel rate = .00223 lbs fuel/sec



Air retained lbs/min =
$$\frac{.00223 \times 60}{.0875}$$
 = 1.53 lbs/min

$$e_s = \frac{1.53}{2.118} = \frac{.617}{}$$

Corrents

This method rave smooth, procise results, and, except for the nacessity of an expensive sampling valve, is considered to be the best of all methods used. It is subject to the same limitations pertaining to sampling valve location, as Method II. The concluding step in this work was a cursory check on the acceptability of a sampling check valve of simple construction and operation, as a suitable replacement for the expensive sampling valve used. For results of these tests, see comments and information under "Sampling Check Valve."



A-PS OL A (contid)

Nethod IV - IJAC Acthod - (Indicated Specific Ale Consent)

ihp - inlicat d horsenover

Ec - fuel heating value,

4 - indicated thornal officiency

btt/lb fuel.

FR - fuel rate - lbs fuel/sec.

(18,900 used for

AC - area of indleator car! - in2

this work)

Subscripts

2 - two stroke

4 - b stroke

This method is based or two main assumptions:

- 1) That in the region of best power operation, the quantity $(F/AE_c + 1)$ of the expression isac = $\frac{2515}{(F/A=c + 1)}$ is
- constant. By measuring (isac) irectly, and evaluating $(P/AL_{C-1})_{\downarrow}$, correcting it for particular 2-stroke operation to $(P/AL_{C-1})_{2}$, isac and e_{s} may be thus evaluated.
- 2) That the combistion characteristics of a 2 strend and histroke cylinder are the same, providing that the following conditions are filled:
 - (a) Premixed charge is used in both cases.
 - (b) The 4-stroke and 2-stroke engines used are of the care, or nearly the same bore and stroke.
 - (c) The 2 encloss must be run at the came misten speci, inlet air temperature, outlet water jacket terminature,



and compression ratio.

In this work, assumption 2(a) has been extended to include injection, and any combination thereof. The discrepancy that occurs in this particular case, where the 4 stroke used a promixed charge, and the 2 stroke an injection fuel system, was accounted for, by using the $(F/A)_2$ results taken from the gas analysis charts, in the correction factor, instead of assuming that at best power the $(F/A)_2$ was the same as the $(F/A)_4$.

The development of the correction factor

$$\frac{(\text{imep} \cdot \forall \mathbf{D} \cdot \mathbf{M})_2}{(\text{imep} \cdot \forall \mathbf{D} \cdot \mathbf{M})_{\downarrow \downarrow}} = \frac{(\mathbf{M}_a \cdot \mathbf{F}/\mathbf{A} \cdot \mathbf{Z}_c \cdot \mathbf{n}_1)_2}{(\mathbf{M}_a \cdot \mathbf{F}/\mathbf{A} \cdot \mathbf{Z}_c \cdot \mathbf{n}_1)_{\downarrow \downarrow}}$$

 $(V_D)_2 = (V_D)_{\downarrow}$ in this case; $(E_c)_2 = (E_c)_{\downarrow}$; $(N)_2 = (N)_{\downarrow}$

 $M_a \cdot F/A = FR$; imep :: AC in both cases.

$$\frac{(AC)_2 \cdot 2}{(AC)_4} = \frac{(FR \cdot \eta_1)_2}{(FR \cdot \eta_1)_4}$$

$$\frac{n_{12}}{n_{14}} = \frac{(FR)_{\downarrow}}{(FR)_{2}} \cdot \frac{(AC)_{2} \cdot 2}{(AC)_{\downarrow}}$$

$$\frac{(F/_{A}Z_{c}h_{1})_{2}}{(F/_{A}Z_{c}h_{1})_{l_{1}}} = \frac{(F/_{A})_{2}}{(F/_{A})_{l_{1}}} \cdot \frac{(FR)_{l_{1}} \cdot (AC)_{2} \cdot 2}{(FR)_{2} \cdot (AC)_{l_{1}}} = \frac{K_{2}}{K_{l_{1}}}$$

 $(F/A)_2$ is the value found by use of sampling valve and analysis charts, as explained in Method II. This correction would not be necessary were both charges premixed. $(F/A)_4$ was .078 for all



4 stroke runs.

K2 = correction factor x K4

(1 sac) =
$$\frac{2545}{2 \text{ or } 4} = \frac{2545}{(F/AE_{cN1})} = \frac{1 \text{ bs air}}{1 \text{ hp hr}} = \frac{1 \text{ bs air}}{1 \text{ hr}} = \frac{1}{1 \text{ hp}}$$

. Air retained =
$$\frac{1}{hr}$$
 $\frac{1}{hp}$. $\frac{h}{hr}$ = $\frac{1}{h}$ $\frac{h}{h}$ $\frac{h}{h}$ = $\frac{1}{h}$ $\frac{h}{h}$ $\frac{h}{h}$ = $\frac{1}{h}$ $\frac{h}{h}$ $\frac{h}{h$

The h and 2-stroke engines used in this work were of the same bore and stroke.

Sample Calculations

2 stroke set at best power. N = 1400, R_s = 1.35 PR = .00219 lbs fuel/sec., T_i = 110°F, T_{water} = 195°F, compression ratio = 6.72. $\rho_s^{NV}D_{r-1}^r$ = 2.51 lbs/min

Indicator card is taken - Area of card = 3.50 in² and Ihp::imep = 13.87 hp.

4 stroke is run; F = 1/400, T_i = 110°F, T_{water} = 195°F, compression ratio = 6.72. FR = .00096 lbs/_{sec}, F/A = .078

Indicator card is taken = Area of card = 3.87 in² and Ihp::imep = 7.67 hp. - Air rate = .0123 lbs/_{sec}.



Indicator cards for h and 2 stroke entires in Figs. ** and ... All (Isac), directly = $\frac{.0123 \times 3600}{7.67}$ = $\frac{2.76}{100}$ lbs air/_{ibp}

$$K_{j_{+}} = \frac{251.5}{5.76} = \frac{11.1}{11.1}$$

K2 = corr. factor • K1

Corr. factor =
$$\frac{(F/A)_2}{.078} \cdot \frac{(FR)_{\parallel}}{(FR)_2} \cdot \frac{(AC)_2}{(AC)_{\parallel}} \cdot 2$$

= $\frac{.08 \frac{1}{4}7}{.078} \cdot \frac{.00096}{.00219} \cdot \frac{3.50}{3.87} \cdot 2 = \frac{.392}{.392}$

...
$$1 \sec 2 = \frac{2515}{x_2} = \frac{6.113}{x_2}$$

...
$$M_{a_{rotained}} = \frac{isac \cdot ihp}{60} = \frac{6.13 \cdot 13.37}{60} = 1.685 \text{ lbs/min}$$

$$e_s = \frac{N_{a_{retained}}}{\rho_s NV_{pr}} = \frac{.672}{2.51} = \frac{.672}{2.51}$$

Comments

The assumptions made in this method appear to be correct in a relative sense, in that this method checks with the other three, within a few percent, as shown in Table I. The method as worked, required an additional equipment, in the form of a sampling valve. This was necessary only in this case where the fuel was injected in one engine, and premixed with air in the



other. Thus, in the general case, no sampling valve would be needed. Results from this method fell on, or near a smooth curve of es vs Rs, and on the basis of results alone, Method IV is as good, or better than any of the other three. However, the equipment needed is usually found only in a laboratory, and the purchase of necessary equipment, solely for purposes of measuring es, would make this method far and away the most expensive of all four methods used.



APPENDIX A (cont'd)

Sampling Check Valve

The investigation of this design was conducted with the object of ascertaining whether or not this type of valve could be used in conjunction with Method III as an alternative to the bulky, expensive timed electrically controlled sampling valve. Tests were short, and inconclusive, but of value.

See Figs. XIV and XIX for location and details of valve.

The valve was mounted rigidly in place, directly in an exhaust port. The distance between the valve opening and the inside cylinder wall was about 1/8". The valve was calibrated statically to open at 30 lbs rage. It was hoped that the dynamic force of the exhaust gases during "blowdown" would be sufficient to open the valve momentarily, and that the spring, and back pressure in the sampling system would operate to keep the valve closed at all other times. The momentary opening of the valve would permit the desired sample to enter the valve, and the closed period would exclude undesirable scavenging air. The sample was led to an Orsat analyzer, where a gas analysis was performed, and an evaluation of es carried out, exactly as in Method III.

Five runs were made with the valve in place, and all were completely successful. Four of the runs were made at various Rs, at 1000 RPM, and defined a smooth curve which was almost



identical with the complete mixing curve, in slope, and curvature, over the range $R_s = 1.0$ to $R_s = 1.8$. See Fig. XIII. At the end of the fifth run, the valve ceased to operate due to clogging, and the test was terminated. Five hours running time was recorded.

The results of the gas enalysis by Orsat, of the samples, were judged to be better than any taken by using the timed sampling valve, since the excess O2 found in all the timed valve samples (discussed under Method II) was lacking, and CO2 and CO content, from analysis, plotted closer to a vertical line on the gas analysis charts, than did the timed valve sample analyses.

A peculiar type of control of the valve performance appeared in the tests. The valve was water-jacketed, and cooled by constant flow, except when samples were taken. This proved necessary due to the fact that when the valve was cooled, the spring developed enough force to keep the valve closed constantly. When the water-cooling was eliminated, the spring apparently got hot enough to cause the spring force to diminish to a point where the valve operated satisfactorily. With the cooling system on, no pressure built up in the line between valve and Creat. When the cooling system was shut down, within a few seconds, the pressure in the line built up to approximately 10 to 15 lbs gage, somewhat in proportion to speed and Rg. If cooling water was turned on, the pressure dropped to 0 gage, in



accordance with the amount of cooling water flowing.

The F/A found using the valve were consistently lower than comparable F/A obtained using the timed sampler of Method III. Typical comparisons were: 1) .0814 to .0850, 2) .082 to .0855, 3) .0803 to .0847.

The engine was set at best power, or very nearly at best power, as described earlier. The sampling check valve gave as a result, one F/A, under .08, the only result of that nature in 28 runs using gas analysis. It had been expected that best power settings would give an F/A of about .078 to .081. The timed valve results were .082 or greater. This was attributed to poor mixing, but the magnitude of the effect was unknown. The sampling check valve helps to evaluate that effect for the region in which the timed valve was located.

Sample calculations are omitted since they are identical with those of Method III.

In order to illustrate "spread" on the Tas Analysis Charts as obtained by use of the sampling check valve, compared against "spread" obtained using the timed sampler, typical runs are illustrated on Fig. XVI. Sampling Check Valve - \triangle , Timed Sampling Valve - \bigcirc



SYMBOLS

In Order of Appearance in Paper.

 o_s - scavenging efficiency = Air retained in cylinder in lbs/min $\rho_s = \frac{1}{r-1}$

es - scavenging density - density of air at pe and Ti

p - pressure

T - temperature - OR

N - revolutions/min

V - volume

r - compression ratio

M_B - rate of mass flow of air

M - mass of substance - used with subscript

 R_s - scavenging ratio = Air delivered to engine in lbs/min ρ s $V_D \frac{r}{r-1}$

Ec - heating value of fuel - bt#/1b

ηi - indicated thermal efficiency

ihp - indicated horse power

F/A - fuel-air ratio

CO2 - carbon dioxide

CO - carbon monoxide

HOH - Potassium hydroxide

CH3NH2 - Monomethylamine

H2SO4 - Sulphuric acid

W - weight of CH3NH2 in 1bs.



WaOH - Sodium hydroxide

(-N) - normality fraction of solution

mH2SO), - molecular weight of H2SO), - 98

MCH3NH2 - molecular weight of CH3NH2

Xcc - Volume of (-N) H2SOh used in sample - in c.c.

Xcccorr - Xcc corrected to end point by titration

and point - Point at which neutralization occurs

Ycc - Vol. of MaOH used in determining end point

 $v_{\rm g}$ - Gas volume bassed thru gasmeter during a sample extraction - ft³

Tg - Temperature of sample passing thru meter - OR

pg - Pressure of sample passing thru meter

 Γ - retention ratio = $\frac{M_{a}rzt}{M_{a}dz!} = \frac{e_{s}}{R_{s}}$

eh - burning efficiency - used in Method I

s - shrinkage factor - used in Method I

Mr - mass of residuals in cylinder - lbs.

Mm - mass of fresh mixture in cylinder - lbs (fresh air only)

Mc - total mass of gases in cylinder - lbs.

mp - molecular weight of residual ras - lbs/lb mol

mm - molecular weight of fresh mixture - lbs/1b mol

mc - molecular weight of cylinder mases - 1bs/lb mol

x - 02 fraction by volume in Mm



 $y = 0_2$ fraction by volume in X_r

s - 02 fraction by volume in Mc

R - universal gas constant = 15hh ft lbs

R - R - ft·lb mols

FR - fuel rate - lbs/sec

AC - area of indicator card - in2

K - constant term F/A. $E_c \cdot \eta_1$ - used with Method IV

Subscripts used

s - indicates point of extraction of expansion sample, except as used with e_s , R_s , e_s . The latter are defined in detail under Symbols.

atm - atmospheric

i - inlet

w - water

d - displacement

e - exhaust

g - gas

exh - exhaust

ret - retained

del - delivered

r - residuals

m - fresh mixture

c - cylinder contents



- a air
- 0 point at which compression sample was taken
- 1 beginning of compression stroke on indicator Hagram
- 2 2 stroke
- 4 1, stroke



APPINDIX B

Included in this section are smooth data sheets, showing results of data and calculations for all points shown on curves of Figs. II thru IX and Fig. XIII.

12345678901234 1516	SPEED R. P. M. /000 /000 /000 /000 /000 /000 /000 /0	R _S 1.795 1.795 1.330 1.250 1.500 1.645 0.996	0.648 0.602 0.584 0.558 0.585	Xcc C.C. 14.0 14.1 14.1	C.C. 10.1 10.2 10.4	C.C. 4.0 4.0	Vg FT ³ 0.324	Tg F	CONCENTR. CH3 NH2 LOS TRACER LOS AIR	X ee c. C.	$\widetilde{X}_{ca}(CoR.)$		V9 3	Tg	, a	S	e _b
12345678901234 1516	1000 1000 1000 1000 1000 1000 1000	1.795 1.330 1.250 1.500 1.500 1.645	0.602 0.584 0.558 0.585	14.0	10.1	4.0	0.324		LOS TRACER	c.C.	C.C	^ ^	3	~			
2345678901234 1516	1000 1000 1000 1000 1000 1000	1.795 1.330 1.250 1.500 1.500 1.645	0.602 0.584 0.558 0.585	14.1 14.1 14.0	10.2	4.0		77			-	C. C.	FT ³	°F	•		
45678901234	1000 1000 1000 1000 1000 1000	1.330 1.250 1.500 1.500 1.645	0.584	14.1	10.4		0770	: //	0.0162	14.0	10.7	3.4	0.511	77	0.673	0.075	0.96
45678901234	1000 1000 1000 1000 1000	1.250 1.500 1.500 1.645	0.558	14.0		7 0	0.330	77	0.0160	14.0	11.1	3.0	0.515	77	0.696	0.075	0.96
56789012314	1000 1000 1000 1000	1.500 1.500 1.645	0.585		1 A A	3.8	0.278	80		14.0	10.8	3.3	0.483	80	0.599	0.075	0.96
6789012314	1000 1000 1000	1.500			10.2	3.9	0.312	82	0.0171	14.0	10.8	3.3	0.558	82	0.593	0.075	0.96
789012314	1000	1.645	0.017	14.2	10.1	4.2	0.310	75	0.0168	14.0	11.0	3.1	0.523	76	0.645		0.96
890121314	1000		0.603	14.0	10.2	<i>3.9 3.9</i>	0.210 6.321	76 76	0.0252	14.0	12.45	1.6 2.8	0.411	76 76	0.624	0.075	0.96
910.12.13.14	1000		0.576	14.0	10.3	3.8	0.240	78	0.0223	14.0	10.8	3.3	0.540	78	0.466	0.075	0.96
10.12.13.14		0.996	0.580	14.0	10.2	3.9	0.325	78	0.0163	14.0	10.3	3. 3	0.745	78	0.462	0.075	0.96
12	1000	0.996	0.550	14.0	10.3	3.8	0.343	. 78	0.0156	14.0	10.5	3.6	0.716	78	0.490	0.075	0.96
13 14 15 16	1000	1.295	0.585	14.0	10.3	3.8	0.286	73	0.0185	14.1	11.0	3.2	0.522	73	0.585	0.075	0.96
14 15 16	1000	1.295	0.595	14.1	10.3	3.9	0.293	74	0.0181	14.0	11.0	3.1	0.541	74	05780.	0.075	0.96
15 16	1000	1.795	0.573	14.0	10./	4.0	0.351	74	0.0148	14.1	11.1	3.1	0.544	74	0.710	0.075	0.96
16	1000	1.795	0.560	13.9	10.1	3.9	0.355	75	0.0147	14.1	11.0	3.2	0.540	74	0.716	0.075	0.96
16		~	•						:		:						
16	1400	1.790	0.580	14.0	10.2	3.9	0.414	73	0.0127	14.0	- 13.8	0.2	0.792	73	0.707	0.075	0.96
17	1400	1.790	0.592	14.1	10.2	4.0	0.407	75	0.0129	14.2	10.9	3.4	0.624	75	0.699	0.075	0.96
17	1400	1.009	0.501	14.0	10.4	3.7	0.460	75	0.0117	14.0	10.8	3.3	0.886	75	0.540	0.075	0.96
18	1400	1.009	0.515	14.0	10.4	3.7	0.460	75	0.0117	14.0	10.7	3.4	0.895	75	0.530	0.075	0.96
19	1400	1.180	0.542	14.1.	11.1	3.1	0.375	75	3 .	14.0	11.5	2.6	0.625	76	0.621	0.075	0.96
20	1400	1.180	0.542	14.1	11.1	3.9	0.375	7 <i>5</i>		20.0	11.1	3.0	0.700	76	0.536	0.075	0.96
22	1400	1.395	0.715	14.0	10.1	4.0	0.318	73	0.0124	14.0	10.75	<i>9.5 3.5</i>	0.633	73	0.529	0.075	0.96
23	1400	1.410	0.66	14.1	: 10.3	3.9	0.293	73	0.0181	14.2	11.00	3.3	0.550	73	0.570	0.075	0.96
24	1400	1.410	0.585	10.0	10.D	00	0.310	7.5	0.0166	14.2	11.0	3.3	0.550	73	0.620	0.075	0.96
25	1400	1.725	0.610	. 14.1	10.1	4.1	0.336	75	0.0155	14.0	10.5	3.6	0.515	75	0.680	0.075	0.96
26	1400	1.725	0.671	14.0	10.2	3.9	0.363	75	0.0145	14.0	10.7	3.4		75	0.645	0.075	0.96
27	1400	1.725	0.680	14.1	10.2	4.0	0.402	77	0.013/	14.0	10.8	3.3	0.665	77			0.96
	1400	1.400	0.600	14.1	10.1	4.1	0.370	73	0.0140	14.0	: 10.8	3. 3	0.650	7.4	0.609	:0.075	0.96
						•			1		t		1				
	Xac	(COR)	= Xcc	- Yee -	(- N) HZ !	DH_		1					•				
;					(- N) HZ	304											:
	THIST	CONCEN	VTOATION	, . C	onstant	10-5	1 Xec Tg		:		•		. ,				
	21666.7	-0110271	1,	`~	1	1		-									
		E (0	1		,	Yq											,
		Xca (Co	R FYHAD	CT V		. U									*		

AIR DELIVERED

LBS/MIN

 0.360
 3.19

 0.345
 3.19

 0.439
 2.36

 0.445
 2.22

 0.390
 2.67

0.411 2.67 0.365 2.94

0.578 1.77

0.58/ 1.77 0.55/ 1.77 0.454 2.275 0.46/ 2.275 0.320 3.160 0.313 3.160

0.323 4.46 0.332 4.46 0.501 2.50 0.511 2.50 0.415 2.91

0.505 Z.91 0.515 3.45 0.514 3.54 0.470 3.54 0.416 3.54

0.353 4.33 0.389 4.33 0.395 4.33

0.428 3.47

 $\Gamma = \frac{e_S}{R_S} = \frac{1-\alpha}{e_b-\alpha S}$

GAS ANALYSIS - COMPRESSION AND EXPANSION SAMPLES Data - METHODS II AND III Compression Sample Expansion Sample

	•				COM	Phasing	n Jampi	e Expo	Phasino	Jample 7										
Run	Speed	R _s	Rot	Fuel Rate	Extraction Angle	" Ernor	Phase Drum Settling	Extraction Angle	Emor	Phase Drum. Setting	Oz in Air	· Oziń Res. Y	Quin chq.	mm	mr	Po	P=	Pe	V ₀	Vc
	RPM			lbs/sec	Deqs.	Dogs.	Deas.	Degs.	Dogs.	Degs.	Co. 0, 100	co2/02/00	0,0,0	lbs/lbmol	Average GM, D+L	psia	psia	psia	in.	in
1	1000	1.790	12.50	.00167	293	8	285	1/1	8	103	0 /21/0	8.6 .6 5.6	3 /192/5	28,93	28,04	33.0	75.0	15.0	19.70	43.89
2	1000	1,790	12,50	.00167	317	. 8	309	///	8	103	-1	8.0 6 6.8		1	. 27.85	61.5	75.0	14,85	12.30	4
3	1000	1.790	12.50	.00167		8	309		8	103		8.8 6 5.6			28.00	61.5	75.0	15,20		
4	1000	1.243	11.75	.00151		8	309		8	/03		80 .6 6.0			27.88	60.8	40.8	15,00		
5	10.00	1.35	11.90	,00154		8	309		8	/03		7.6 .6 6.6			27.75	61.8	61.8	15.00		
6	1000	1.505	12,10	.00158		8	309		8	103		7.4 .6 6.4			27.70	62,8	63.8	15.00		
	. 1000	1.66	12,30	.00163		8	309		8	/03		7.4 .6 6.6			27.67	62.8	67.3	15.00		
8	1000	1.795	12,50	.00167		8	309		0	103		6.6 . 6 3.7	. t		27.94	65.7	69.7	14.90		ļ
4	1000	1.33	11.90	. 00154	'	0	309		δ σ	103		6.2 .6 4.2			27.70	64.3	64,2	14.90		
10	1060	1.25	11.75	.00151		8	309		8	103		6.5 .6 4.8			27.72	64.7	61.7	14.90		
11	1000	1.645	12.30	.00163			309		8	/03		8.5 . 6 4.2			28,26	64.7	67.7	14.90		
12	1000	1,295	10.80			0	, 309 309	į	0	103	e. The	7.6 .6 4.8			28.05	62,7	54.7	14.80		ł
13	1000	1.795	11.90	.00153		8	309		9.	103		6.9 .6 4.9 83 .6 5.9			27.88	64.5	58.5 63.5	14.70		
14	1400	1.652	15.7	.00237		0	306		11	100		7.8 .6 4.2			28.08	64.7	74.7	15.20		
1/	1400	1.505	15.35			11	306		11	100		7.4 .6 4.2			28.04	63.7	72.7	15.20		
10	1400			.00220		//	306		. "	100		7.3 ,6 4.2			28.04	62,7	68.7	15.20		
18	1400	1.790	16.10	.00246		11	306		. 11	100		7.2 .6 4.2	10 19.7 4		28.05	66.7	77.7	15.10		
19	1400	1.009	13.50	.00188	. 4	7/	306	V	11	100		6.6 .6 3.4	1.2 18.2 1.2		27.95	60.5	59.5	14.70		
20	1400	1.18	14.10	100202		//	306		11	100	V	7.0 .6 4.0		V .	28.00	61.5	64.7	14.70		
21	1400	1.40		.00223		11	306		11	100		7.6 .6 6.0	.6 19.0 -		. 27.77	68.5	60.5	14.90		Y

mark!

	Chart				Curre														
Ti		Mm Mm/M	Mr/Mc	mc	Expansion	Tr	Tr'	T	T,	To	1/ May	Po/Pe	Vo/Vc	Ti/To	mc/mm	es	es	es	e,
٥F	6+ 10+4 -			lbs/ lb mol		°R	°R	°R	°R	°R	_		_		<u> </u>	cas	G+M Chart	D+L Chart	Average Chartes
110		96 .915	.085	28.83	5.20	3270	2145	1845	678 672	810 978	.915 .915	2.20	. 45	.705	.998	.636	. 666	.658	.662
	.0835.084 .0	850 .921	.079	28.86		3260	2155	1855	672	966	.921	4.05		,590	,999	.619	.662	,658	. 660
		141 .875 28 .886	.125	28.82 28.76		32/0	2230	1930	741 719	1068	.875	4.055		.546	1.00	.530 .554	.282		. 585
		93 .914	. 086	28.77		3155	2165	1865	681	985	.914	4.19		.579	999	.615	.608	608	. 608
	.0889 .0886 .0	814 .924 66 .939	. 076	28.80 28.86		<i>3145 3210</i>	2/30.	1830 1855	665	965	924	4.19	:	.579 .591 .598 .594	999		. 614	.614 .660	. 614
	.6882 .0877 .0	705 .933	,067	28.85		3170	2165	1865	656	960	.933	4.31		.594	.999	.665	.591	594	. 593
	.0885.0880 ,0	983 .910 830 .924	,090	28.79		3/60	2185	1885	688 672	985	.910	4.34		.567	,995	.621	. 577	.575	.576
	.085 .1	67 .858	142	28.79		3240	23/0	2010	175	1120	.858	4.21		.508	999	.515	.525	,525	. 525
	.086 .087 .1	22 .890 714 .931	.110.	28.88 28.88		3200	2235	1935	720	1060	.890 .931	4.39		.578	1.00	. 588 . 690	.606	.600	.60 3 .666
	. 0835.084 . 0	66 .937	7 .663	28.88		3260	2145	1845	650	946	.937	4.26		.602	.999	.670	. 679	.675	.677
	.085 .0845 .0			28.88 28.93		3240	2150	1850	667 687	967 991 951 1090	.924	4.19		.590	1.00	. 637	.644	.647	
	.685 .085 .0	66 .938	.062	28.89		3230	2145	1845	649	951	910	4.41		.598	1.00	,693	.617	.694	.694
	.086.085 .1	54 ,866 05 : .965	134	28.74 28.86		3220	2190	1930	752	1010	,866	4.11		.522	1.00	.518	.526	.578	.529
1	.0875 .0875 . 1		- and the second	78.79	1	3170	2200	1900	697	1040	.905	4.59	\	.548	,99.6	,645	.570	,617	.617

$$\frac{M_{\Lambda}}{M_{m}} = \frac{x-z}{z-y} \cdot \frac{m_{\Lambda}}{m_{m}}$$

$$\frac{M_{m}}{M_{c}} = \frac{1}{1+\frac{M_{\Lambda}}{M_{c}}}$$

$$\frac{M_{m}}{M_{c}} = \frac{1}{1+\frac{M_{\Lambda}}{M_{c}}}$$

$$\frac{M_{\Lambda}}{M_{c}} = 1-\frac{M_{m}}{M_{c}}$$

$$\frac{M_{\Lambda}}{M_{c}} = 1-\frac{M_{m}}{M_{c}}$$

$$\frac{M_{\Lambda}}{M_{c}} = \frac{1}{1+\frac{M_{\Lambda}}{M_{c}}}$$

$$\frac{M_{\Lambda}}{M_{c}} = \frac{1}{1+\frac$$

Formulae

I.S.A.C. Method. - METHOD IV

Formulae:

Correction factor =
$$\frac{[F_A]_2}{O78}$$
 (Area Card)₂ (FR)₄. 2

(Isac)₄ = $\frac{2545}{[F_A]_4}$ = $\frac{(arr note in |bs|./1)}{hr/(ihp.)}$
 K_4
 $[F_A : E_c : \gamma_i]_2$ = $[F_A : E_c : \gamma_i]_4$. Corn factor.

 K_2
 K_4

(Isac)₂ = $\frac{2545}{K_2}$

Air retained = $\frac{2545}{K_2}$

Air retained = $\frac{2545}{K_2}$

SAMPLING CHECK VALVE DOTA

Run RPM Rs Rot. Fire! BNVA CO2 O2 CO FA FA Manet Maket. Es

- - | bs/sec | bs/min lo by vol. loby vol. loby vol. G+M. D+L. | bs/min | bs/min G+M.

1 1000 1.02 10.8 .00132 1.782 10.0 .4 6.0 .0811 .0817 .976 .973 .548

2 1000 1.505 12.1 .00158 1.785 10.2 .3 6.4 .0815 .0820 1.162 1.158 .652

3 1000 1.765 12.4 .00166 1.810 11.0 .0 4.8 .077 .077 1.795 1.295 .716

4 1000 1.258 11.80 .001515 1.81 10.2 .2 6.2 .0812 .0820 1.12 1.11 .620

5 1400 1.495 15.30 .00228 2.51 10.6 .3 6.1 .0800 .0805 1.71 1.70 .680

4. Stroke Comparison Runs Data (I.S.A.C. Method)

RUN						Corr. Factor									
	RPM	in. Hg	in. Hy	in. Hg	ot-	0.7.7	oF	in. H20	1bs/sec	Ibs/	Degrees	oF		16s. Fuel	oF
1	1400	29.9	-1.0	40.2	110	0.975	70	8.4	.0/23	.00096	24	. 195	8.3	.078	70
2	1000	29.9	-1.0	+0.2	110	0.975	70	5.3	.00985	.000769	20	195	7. 3	.078	70



APPENDIX C

Original data sheets for this work are included in the M.I.T. copy of the thesis, and are on file in Sloan Laboratory.

See Appendix of for smooth data sheets, including all points that were considered to be free from any of the causes contributing to inaccuracy or poor precision.



APPENDIX D

Bibliography:

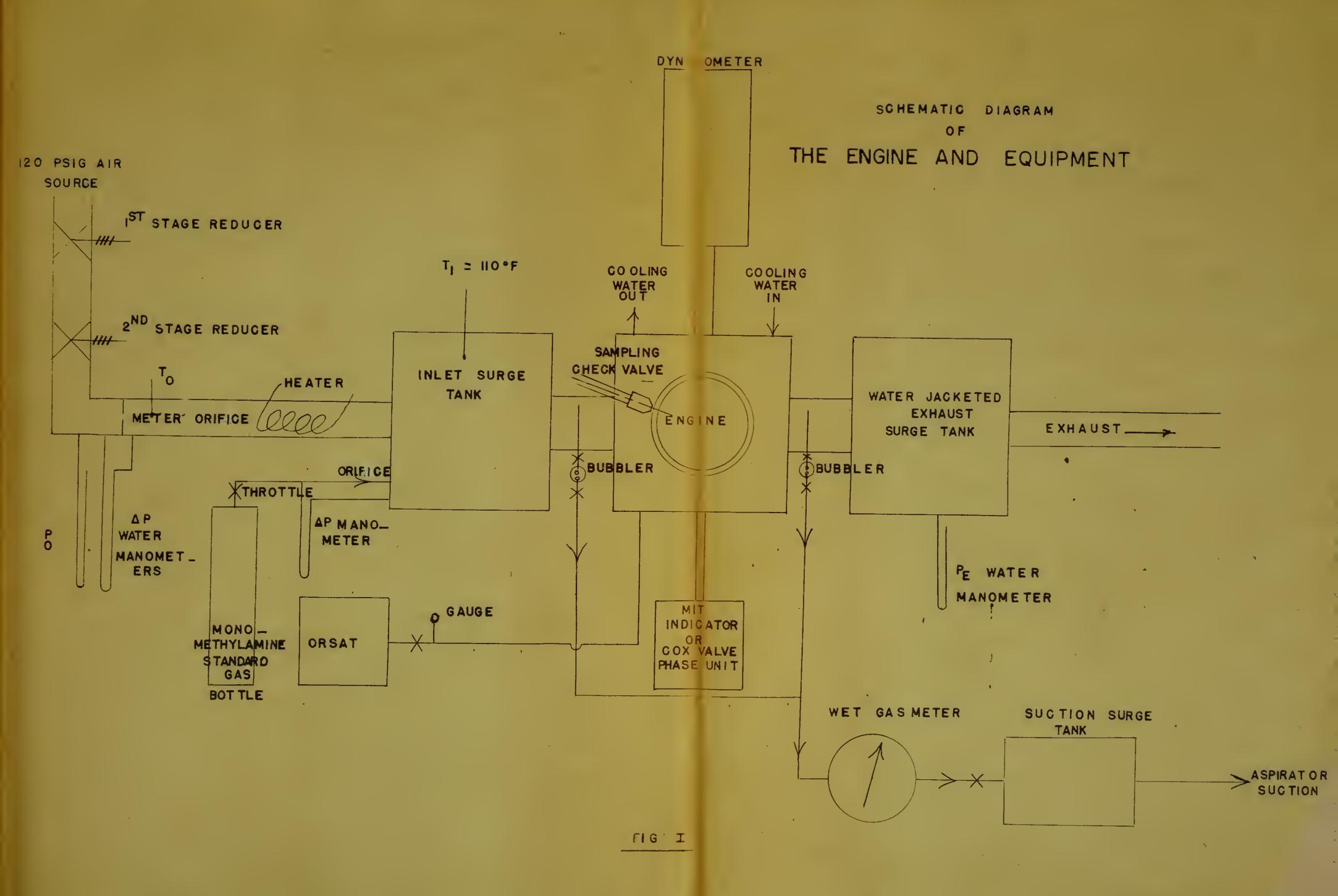
Texts or Pamphlets

- A. The Internal Combustion Engine Taylor and Taylor
- B. Report on 2-Stroke Engines C. F. Taylor
- C. NACA Technical Notes #838 The Tracer Gas Method of Determining the Charging Efficiency of Two-Stroke-Cycle Diesel Engines - Schweitzer and DeLuca

Figures or Curves

- 1) Analysis of Exhaust Jases from a Spark Ignition Engine D'Alleva and Lovell "Relation of Exhaust Gas Composition
 to Air-Puel Ratio" SAE Journal, Vol. 38, No. 3, March, 1936
- 2) Relation of Constituents in Normal Exhaust Gas to Fuel Air Ratio Gerrish and Meems Figure 3 of NACA Wartime Report October, 1943. See NACA Report #757
- 3) Conditions at End of Expansion, Constant Volume Fuel-Air Cycle Goodenough and Baker "A Thermodynamic Analysis of Internal Combustion Engine Cycles" Univ. of Illinois Eng. Exp. Sta. Bulletin 160, 1927







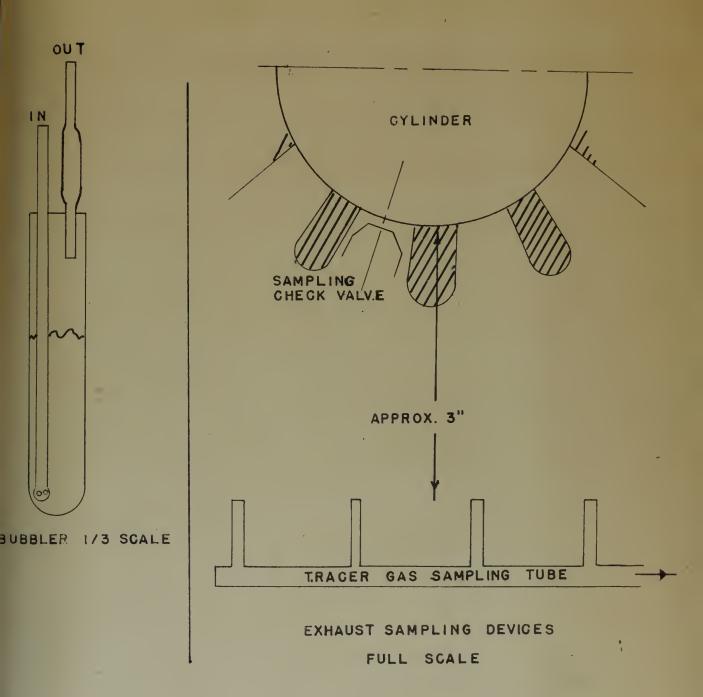
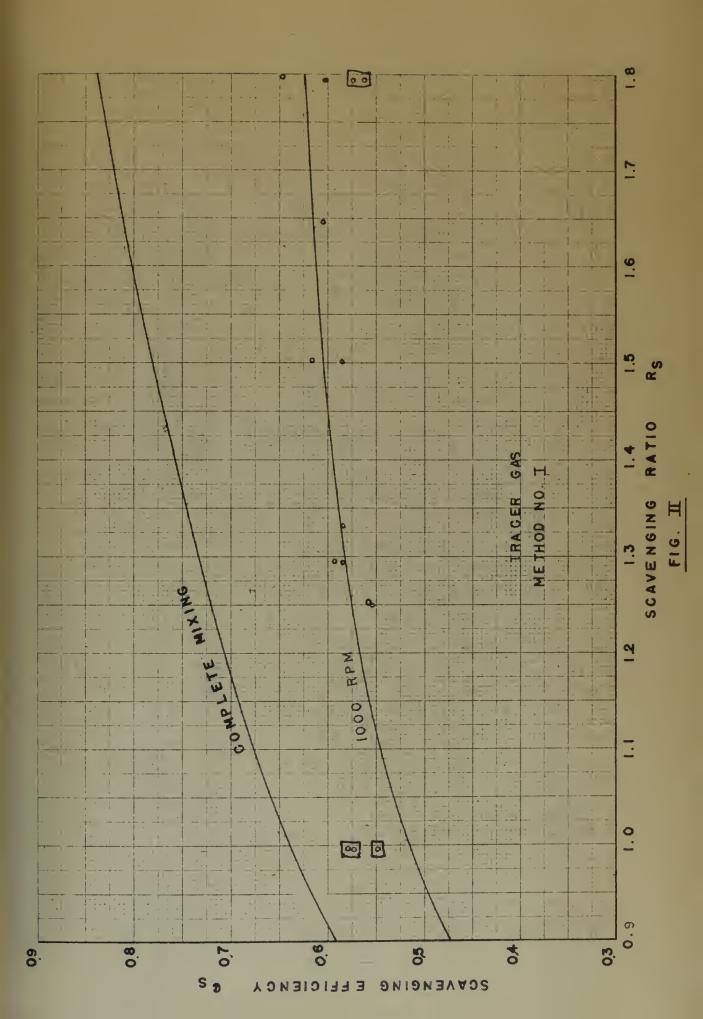
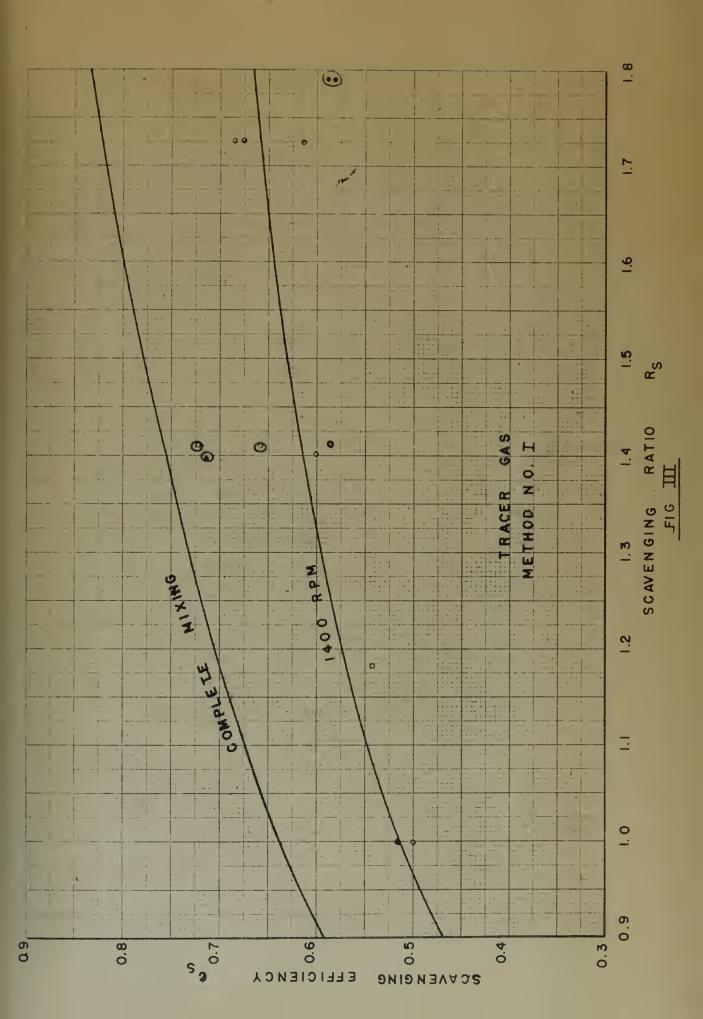


FIG. IA

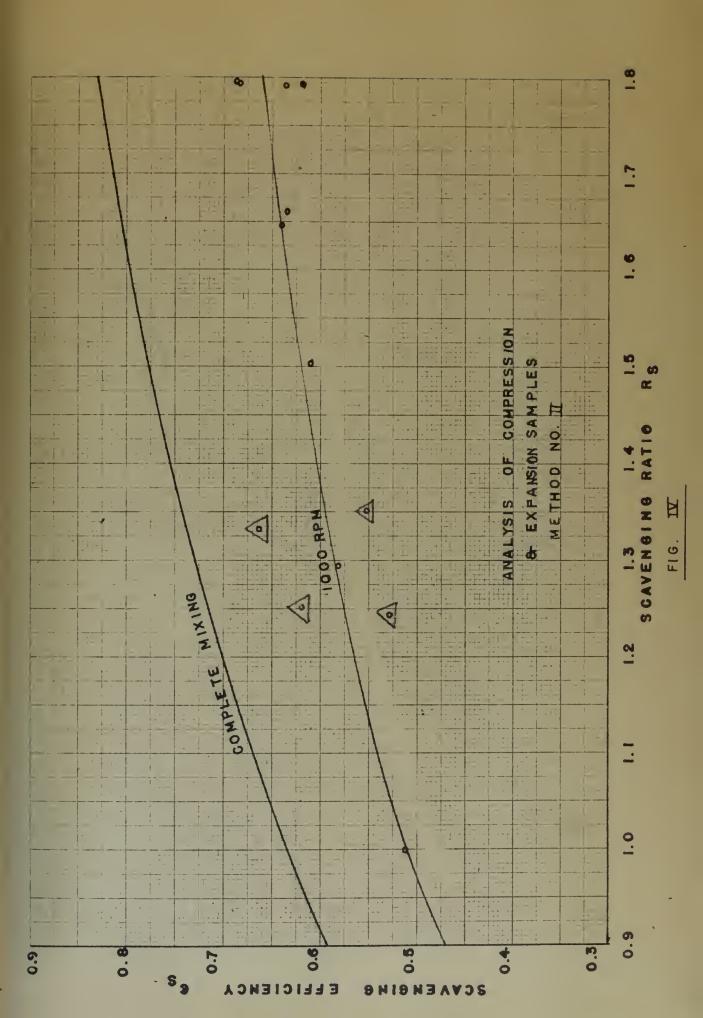




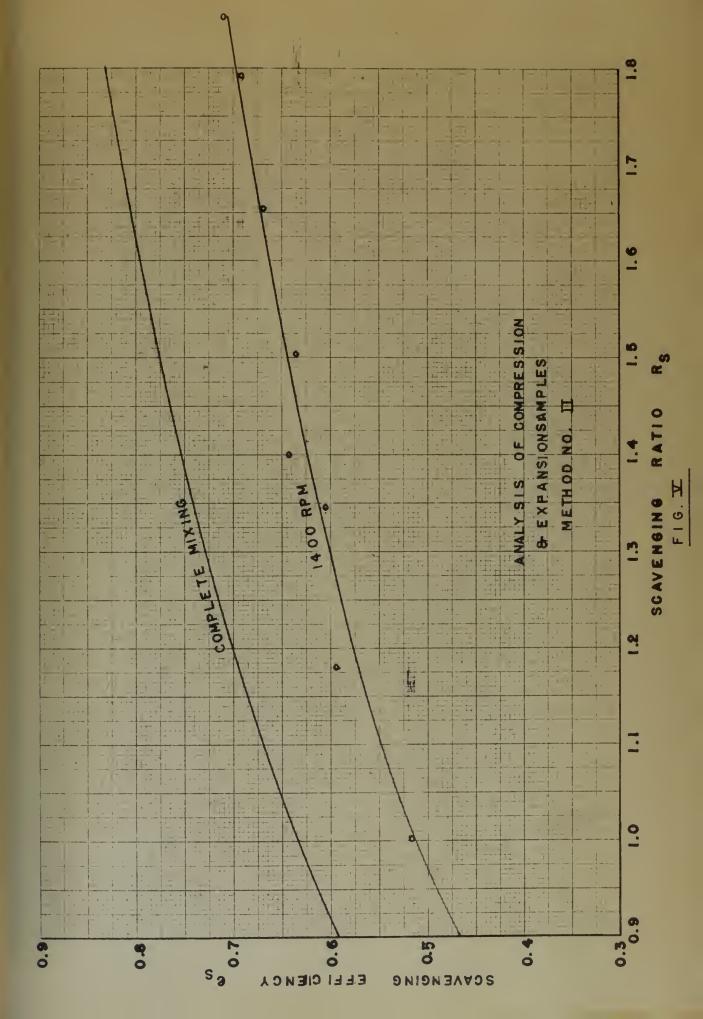




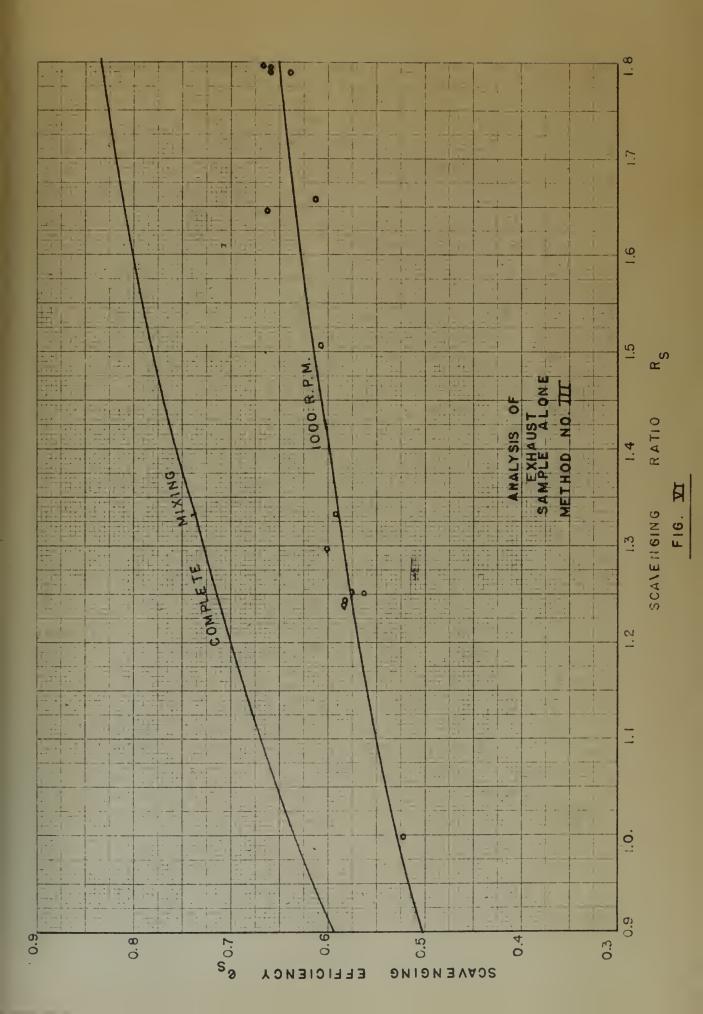




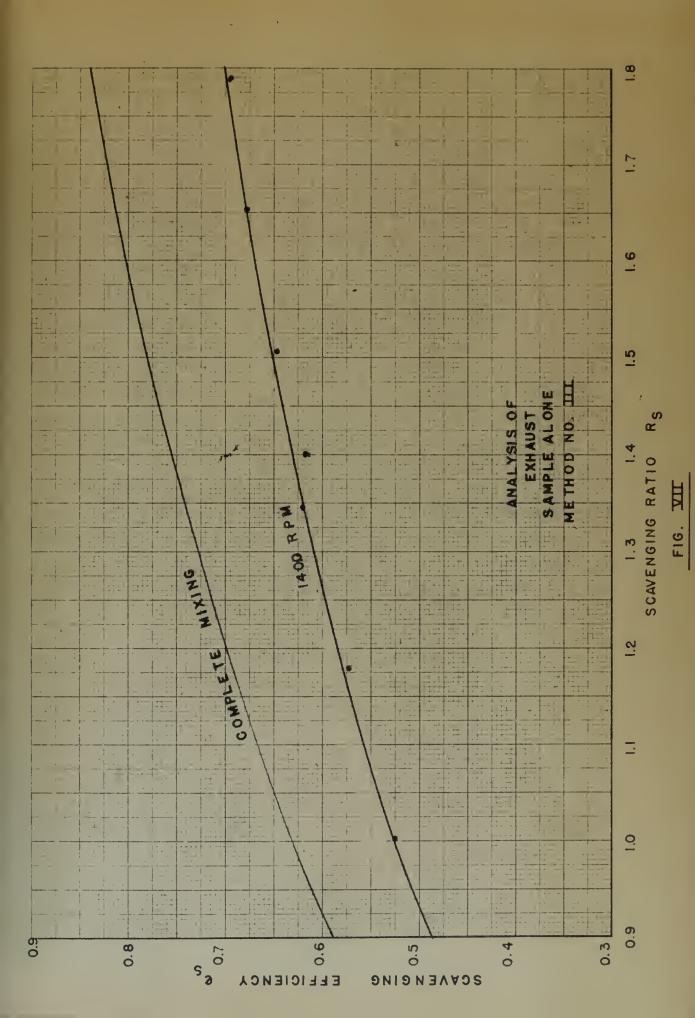




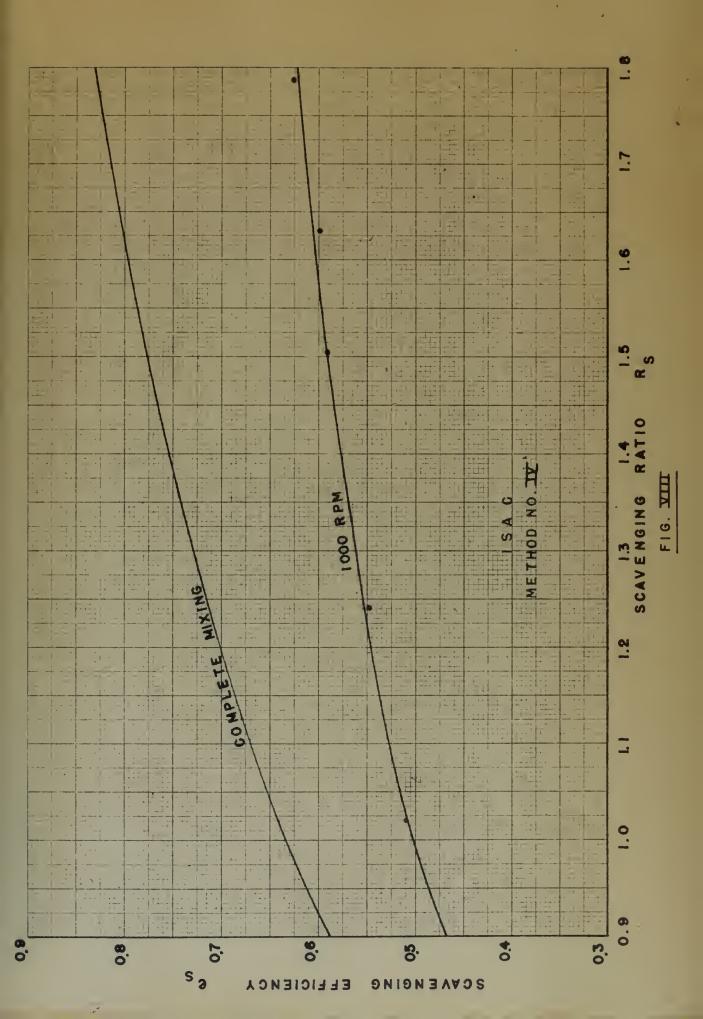




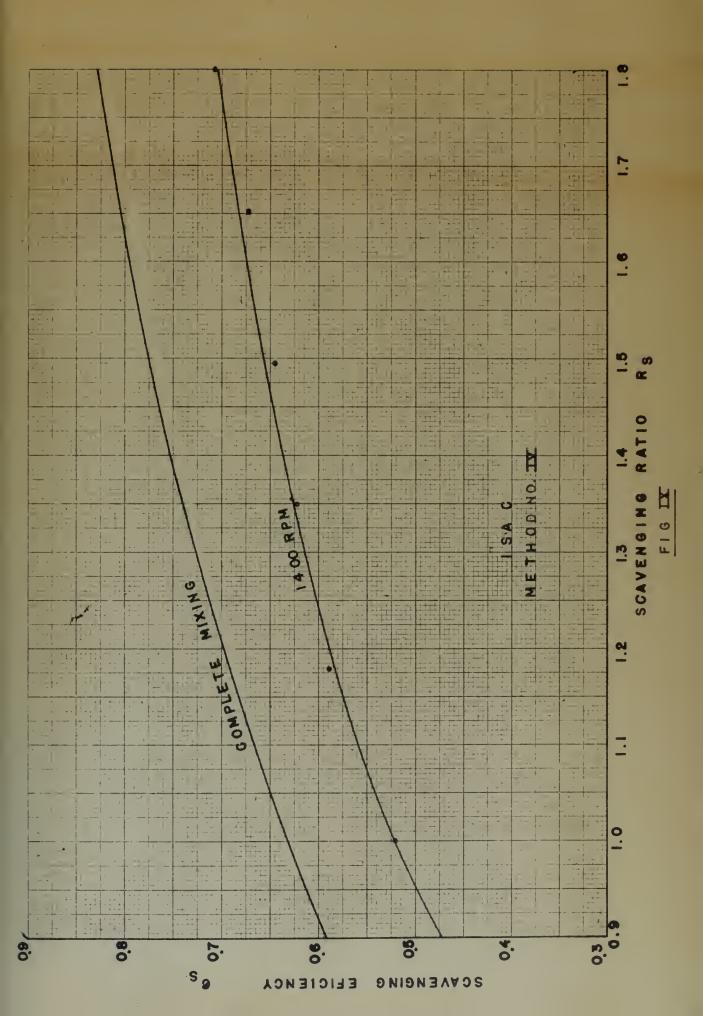




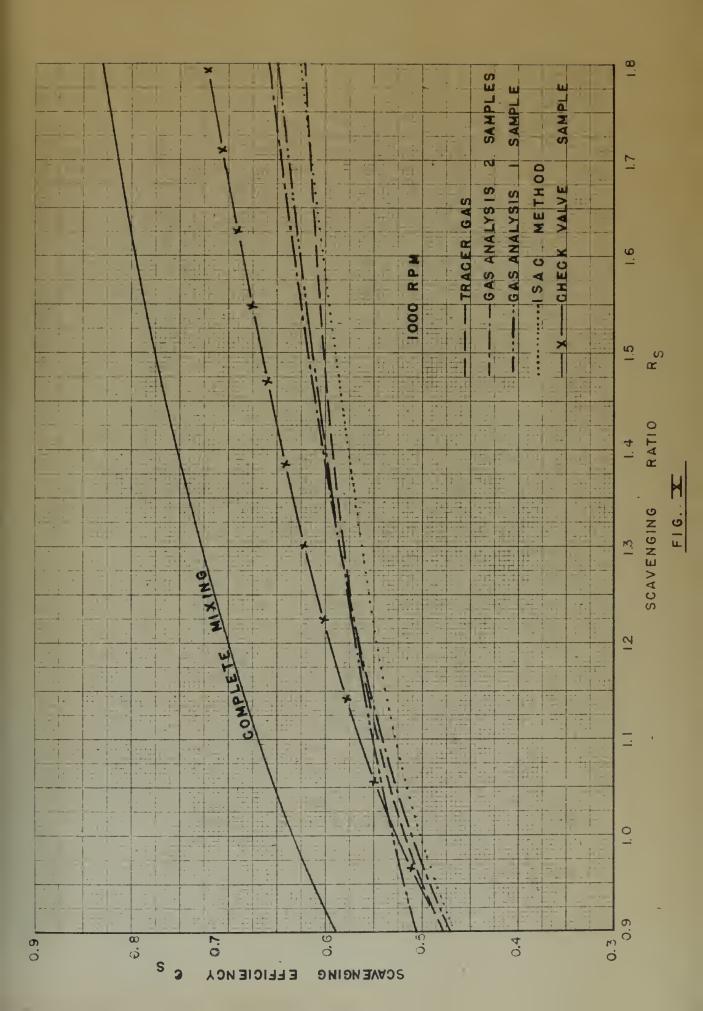




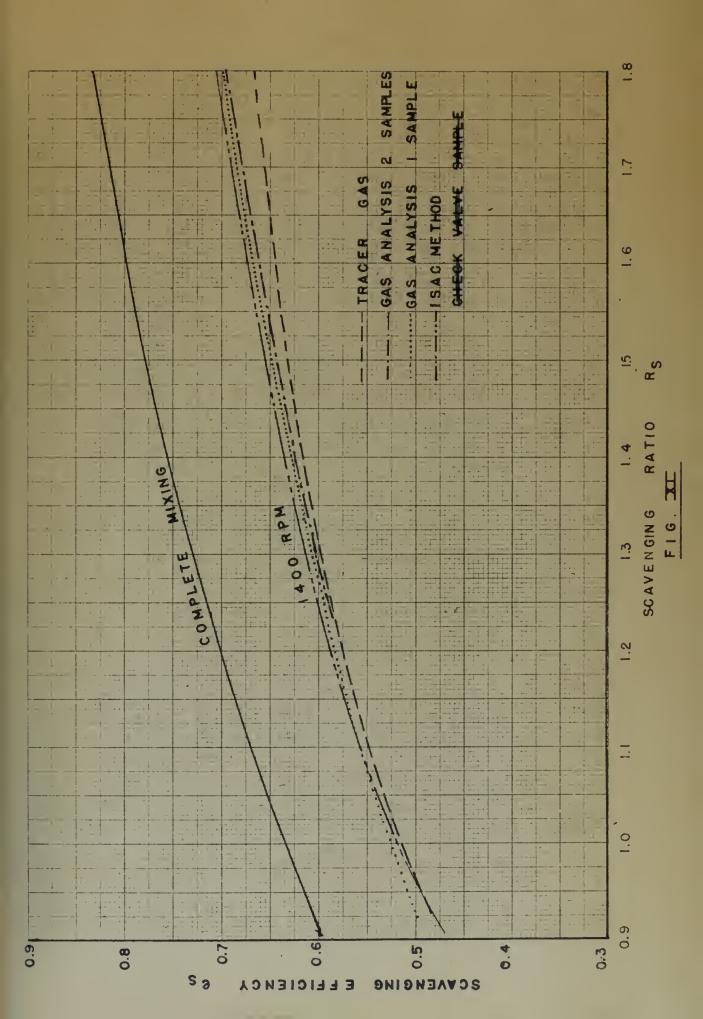




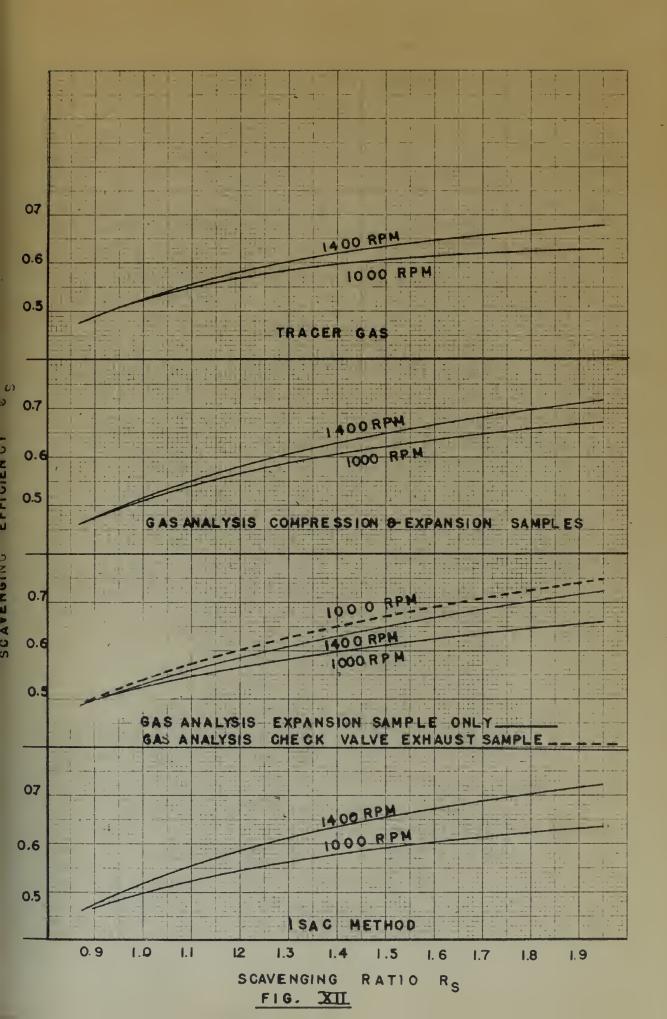






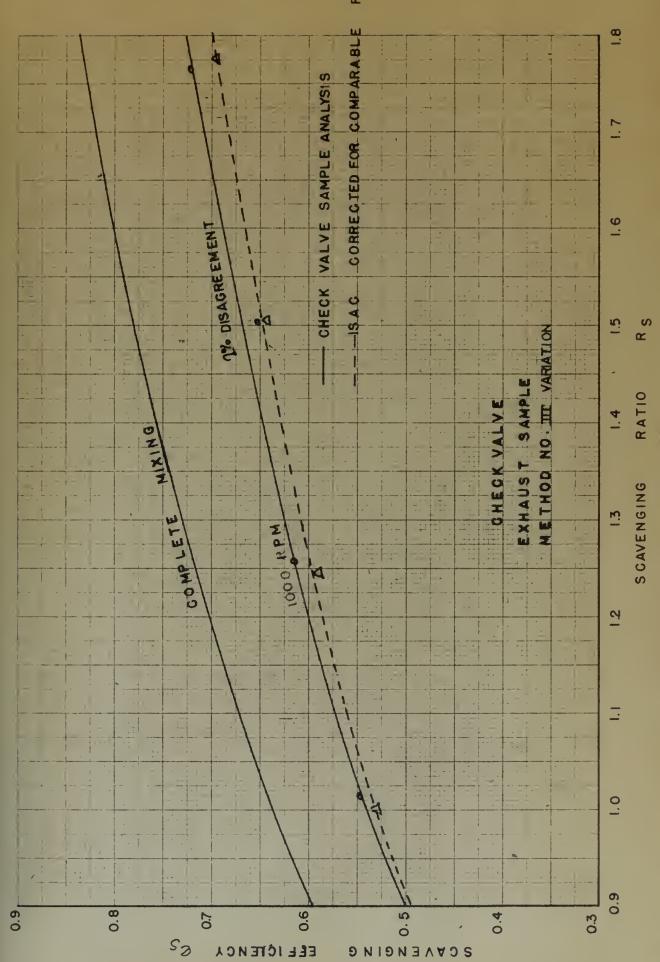






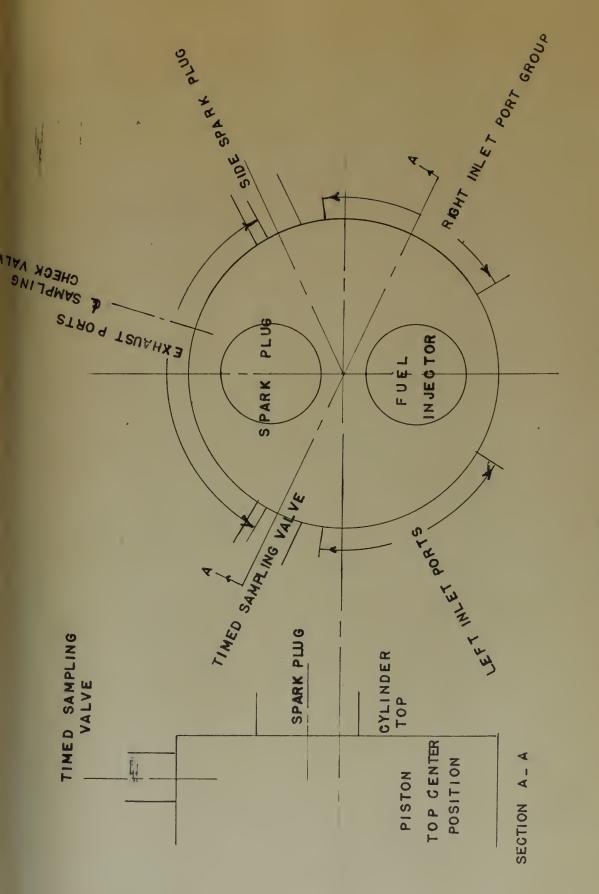






FI G.





CYLINDER HEAD DETAILS

FIGXIV



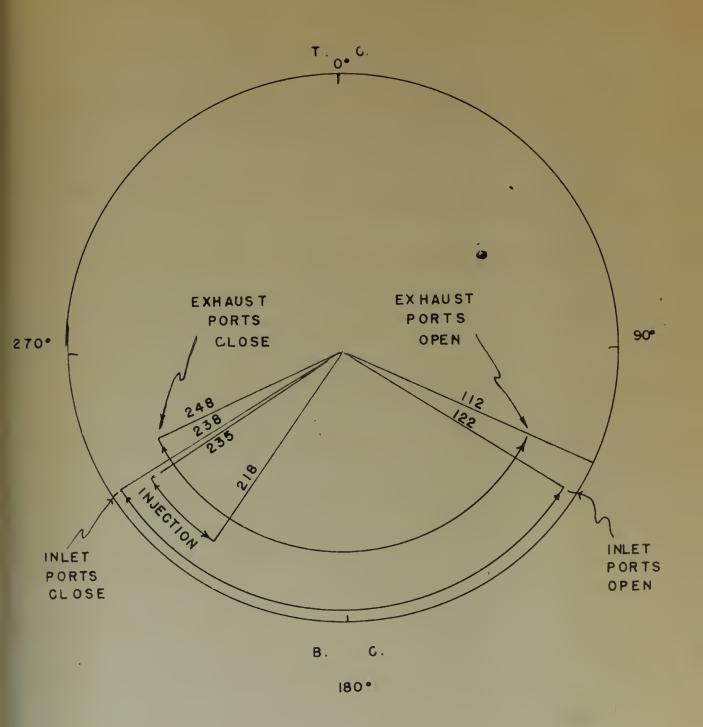
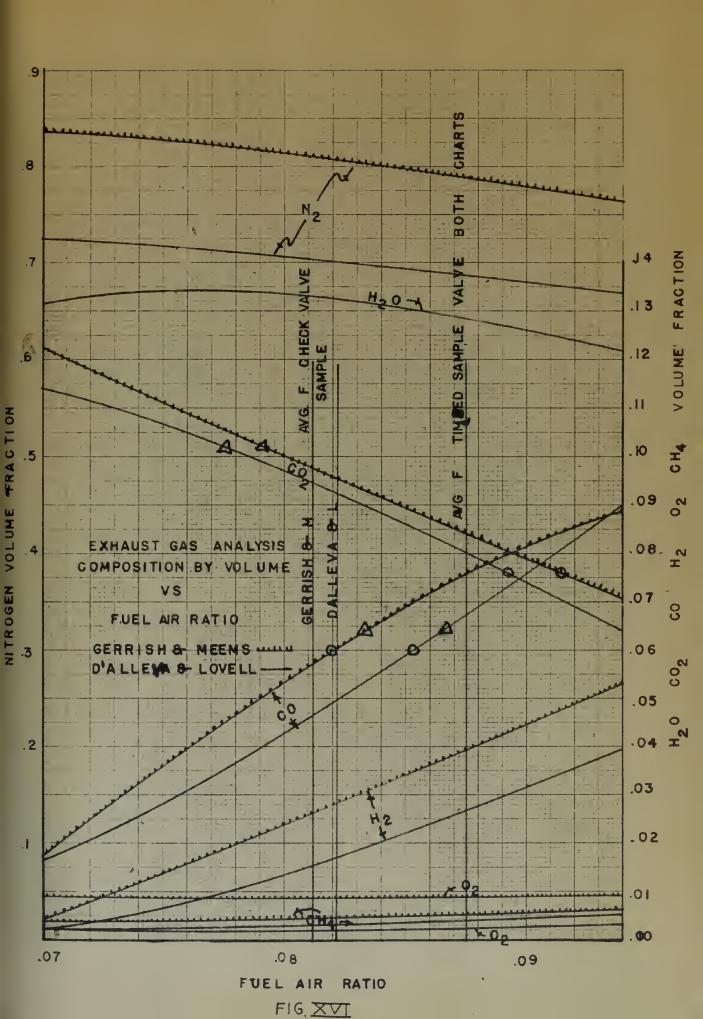
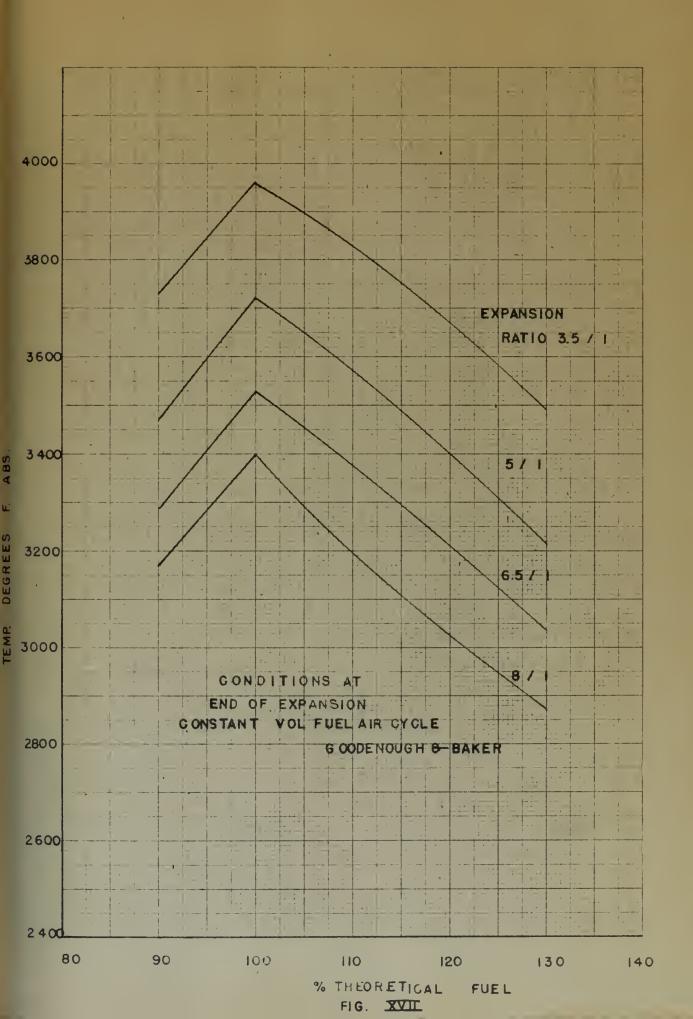


FIGURE XY
TIMING DIAGRAM ~











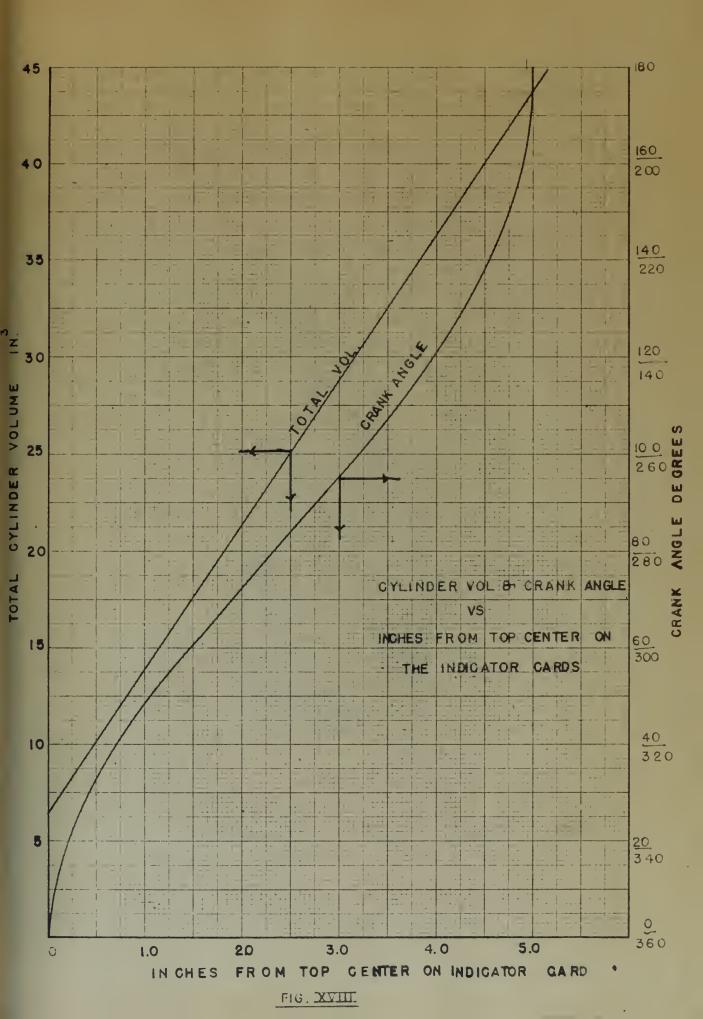






FIG XIX (4) SAMPLING CHECK VALVE (ASSEMBLED)

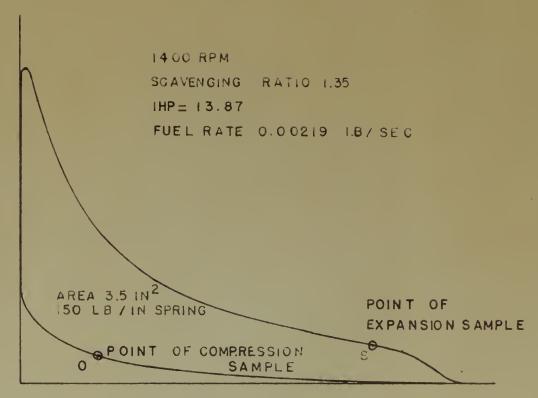






Fig. XIX (B) SAMPLING CHECK VALVE (LAID Oct)





INDICATOR CARD FOR TWO STROKE CYCLE

FIG. 20



I 400 RPM

IMEP II6.I PSIA

FUEL AIR RATIO 0.078

IHP= 7.67 HP

AIR RATE 0.0123 LB/SEC.

FUEL RATE 0.00096 LB/SEC.

I SAC 5.76 LB/IHP HR.

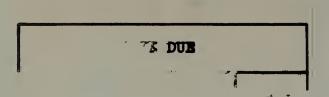
K₄ = 430

AREA 3.87 N²

I50 LB/IN SPRING

INDICATOR CARD FOR FOUR STROKE CYCLE

FIG. XXI





thesi6
Measurement of scavenging efficiency in

3 2768 002 10181 8
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